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The Structural Design, Simulation, and Analysis of a Wave Impact Simulation Device for the Component Flooding Evaluation Laboratory by

## Larinda Nichols

A thesis<br>submitted in partial fulfillment<br>of the requirements for the degree of<br>Master of Science in the Department of Nuclear Science and Engineering<br>Idaho State University

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## COMMITTEE APPROVAL

To the Graduate Faculty:
The members of the committee appointed to examine the thesis of Larinda Nichols find it satisfactory and recommend that it be accepted.

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Major Advisor

Dr. Chad Pope,
Committee Member

Dr. Brian Williams,<br>Graduate Faculty Representative

To my children, Dennis and Ayla...
May you never forget how we got here. There are no words that can describe my love for you. Thank you for helping mommy achieve her goals.

To my own mother...
What a long strange trip this has been. I couldn't have done it without you.

To my Sammy...
You are truly my Atlas.

## ACKNOWLEDGEMENTS

Dr. George Imel for his countless hours of patience with me. I am in constant awe of his humble brilliance. I want to be Dr. Imel when I grow up.

Dr. Chad Pope. You make every student feel uniquely important even though you are the busiest man I know. Thank you for teaching me the importance of balance. Your leadership and encouragement has helped me to accomplish more than I ever thought possible.

## VITA

Larinda is the youngest of her four siblings, all born and raised in Southeast Idaho. On the day Larinda was born, her older brother Myron crawled under the kitchen table and cried. He was devastated that his mother had given birth to yet another girl and not the little brother he had wanted so badly. So like all good brothers, he decided instead to teach Larinda everything a boy should know. How to ride a bike. How to ride horses. How to snow machine and 4-wheel. He even taught Larinda how to wrestle. The elementary wrestling coaches were sure it was only a joke when she showed up to practice in her black spandex shorts with the pink ruffle. But Larinda made her brother very proud when she took $2^{\text {nd }}$ place in the wrestling tournament that year. When she entered high school, Larinda decided it was time to support her wrestling buddies in a "less awkward way," so she turned in her wrestling outfit for a pair of pom-poms.

After high school, Larinda went on to receive a degree in Graphic Communications from Idaho State University. She worked in Pocatello for three years at a local magazine company until the birth of her first child Dennis, followed by his little sister Ayla two years later. Larinda stayed at home and raised the two children until the youngest started kindergarten. But rather than going back to work, Larinda realized that nothing could be as challenging as raising two toddlers. So, she came back to ISU to embark on what would become a seven-year adventure in Nuclear and Mechanical Engineering.

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## LIST OF ABBREVIATIONS

| BWR | Boiling Water Reactor |
| :--- | :--- |
| CFD | Computational Fluid Dynamics |
| CFEL | Component Flooding Evaluation Laboratory |
| FEA | Finite Element Analysis |
| INL | Idaho National Laboratory |
| LWRS | Light Water Reactor Sustainability |
| NRC | Nuclear Regulatory Commission |
| PET | Pressurized Evaluation Tank |
| PWR | Pressurized Water Reactor |
| WISD | Wave Impact Simulation Device |

ABSTRACT<br>Structural Design, Simulation, and Analysis of a Wave Impact Simulation Device<br>for the Component Flooding Evaluation Laboratory<br>Thesis Abstract--Idaho State University (2018)

A Wave Impact Simulation Device (WISD) has been designed for the Idaho State University Component Flooding Evaluation Laboratory (CFEL). The WISD will be used to test the resilience of components and component sub-assemblies when subjected to a 10 ft . by 10 ft . jet of water using a pneumatically driven, ten level water system. The ten channels shall be constructed of $10 \mathrm{ft} . \mathrm{x} 10 \mathrm{ft}$. x $3 / 8 \mathrm{in}$. thick, A36 steel plates supported by interspersed $1 / 4 \mathrm{in}$. steel rod supports. The gates of the prototype shall be $1-3 / 4$ in. keyed shafts with $3 / 8$ in thick steel, rectangular plates spanning the width of each channel. Electromagnets will be used for the gate mechanism to achieve a $90.8 \mathrm{rad} / \mathrm{s}$ angular velocity and 90 -degree travel time of 0.02 seconds.

Key Words: wave impact simulation, component flooding evaluation, tsunami research

## 1. INTRODUCTION

## "To stay the course, sometimes you have to make waves." <br> (Author Unknown)

In 2017, the United States obtained approximately $20 \%$ of the nation's electrical energy from nuclear power. Of the 99 licensed to operate nuclear power plants, 65 are Pressurized Water Reactors and the remaining 34 are Boiling Water Reactors (World Nuclear Association 2018). Both require large quantities of water. Not only is water used to transfer the heat energy from the reactor to the steam turbines, but it is also essential for cooling and moderating the reactor core. Thus, it is reasonable that these nuclear plants were built near large bodies of water, where a continuous and plentiful supply of coolant is readily obtainable.

The drawback to this strategy was never more apparent than on March 11, 2011 when tsunami waves of approximately 50 feet eventually led to the meltdown of three reactor cores at the Fukushima Daiichi Nuclear Power Plant. An estimated 940 PBq $(25,405,405 \mathrm{Ci})$ of radioactivity was released in just six days (World Nuclear Association 2017). It was one of the most massive nuclear accidents occurring to date, and it dealt yet another sharp blow to an already struggling nuclear industry.

Within four months, the United States Nuclear Regulatory Commission (NRC) had established a "Near-Term Task Force" to review its current processes and regulations. A classified report, also dated four months after Fukushima, identified 34 nuclear plants considered to be "at heightened risk of flood damage due to upstream dam failures" (Perkins, Bensi, Philips, and Sancaktar 2011). The true potential of flood damage was now realized. Complex testing on individual components and component subassemblies under various flooding conditions was essential in circumventing future nuclear disasters.

In 2015, a partnership between the U.S. Department of Energy's Light Water Reactor Sustainability Program (LWRS) and Idaho State University's Component Flooding Evaluation Laboratory (CFEL) was formed to improve flood risk analysis at nuclear power plants. The experimental program would develop fragility curves and reliability models based on water rise, spray, and wave impact testing. Risks associated with both internal and external flooding to nuclear power plants would then be identified and used to better quantify safety margin.

Currently, the CFEL team is collecting valuable experimental data using a portal evaluation tank, referred to as the PET. Commercial doors, both hollow and steel, are tested for failure using the PET's unique structural capabilities. Data on flowrate, pressure, mode of failure, and fracture patterns all contribute to the development of the fragility curves and reliability models necessary for a more accurate risk assessment. Additionally, the relationship between water height and door strength observed by the CFEL team has already provided previously unknown information to the LWRS Program.

A second structure, known as the Wave Impact Simulation Device (WISD), is also proposed for component testing at CFEL. The device will simulate impact forces equivalent to those of a full-scale tsunami wave by way of rapid water acceleration. Component resilience, response, and/or failure when subjected to these forces will be observed, and the data extracted from each experiment will be used to further advance the analysis models being developed by the team. The following thesis shall focus on the development of a physical prototype for the WISD.

## 2. BACKGROUND

The work performed in this thesis is a continuation of research begun in 2015 by Gregory David Roberts. In his thesis "Research and Development of a Wave Impact Simulation Device for the Idaho State University Component Flooding Evaluation Laboratory," Roberts successfully identified the concept for a high velocity jet to satisfy the requirements of the project. Namely, the velocity and momentum of a 20 ft . wave must be replicated in a near vertical 10 ft . by 10 ft . section of water for impact testing (Roberts 2016).

Roberts' approach to tsunami wave celerity was to use shallow water wave equations. He justified his approach by stating the governing characteristics of tsunami waves during impact include an inundation and turbulence that cannot be described by generalized wave theory. Under this assumption, Roberts found that a 20 ft . wave would have a maximum horizontal fluid velocity of $25.4 \mathrm{ft} / \mathrm{s}$. Obtaining the constant $25.4 \mathrm{ft} / \mathrm{s}$ velocity was made a key requirement for the WISD.

Using the computational fluid dynamics (CFD) software Flow-3D, Roberts was able to simulate resultant wave profiles for various device geometries, configurations, and driving mechanisms. Ten design approaches were documented in his work. His conclusion was that the most promising design for delivering a near vertical, $10-\mathrm{ft}$. by $10-\mathrm{ft}$. wave, at 25.4 feet per second was to use a system of plates coupled with rapid response gates. Roberts WISD concept is illustrated in Figure 1 and Figure 2.

Roberts concluded his research by recommending the most viable displacement mechanism for accelerating the static reservoir section of water to its required velocity before exiting the plate section would be a pneumatic air system. Adjustments to the angle of entry and length of exit conduit were also recommended as they were found to have a substantial impact on the velocity and flow profile.

The following thesis includes research, calculations, simulations, and analyses on the many iterations undertaken by the WISD team to design a device meeting all project requirements as established by Roberts.


Figure 1. WISD concept developed by Gregory Roberts at $t=0$ seconds (Roberts 2016).

Time Frame: $\mathbf{1 . 2 1}$


Figure 2. WISD concept developed by Gregory Roberts at $t=1.21$ seconds (Roberts 2016).

## 3. LITERATURE REVIEW

The purpose of the WISD is to subject various components and component subassemblies to worse-case, wave impact scenarios, then observe their response. The device must be of sufficient strength to withstand the forces generated by a substantial volume of water in both stationary and turbulent flow conditions, but also sleek enough to minimize interference with the desired wave formation. A successful solution to these requirements would first require preliminary research. A literature review was performed to glean information from other current laboratory wave generation methods.

### 3.1 PAST AND CURRENT TSUNAMI SIMULATION MODELS

Historically, analytical approaches to tsunami research date back to as far as 1896 after the Meiji Great Sanriku Tsunami claimed 22,000 Japanese lives (Shuto and Fujima 2009). Wave heights up to 30 ft . were recorded as far as the Hawaiian Islands (Yuichiro 2001). Rudimentary countermeasures for evacuation where introduced in 1933 when the largest recorded normal-faulting earthquake in history produced the Showa Great Sanriku Tsunami (Okal, Kirby and Kalligeris 2016). By 1941, tsunami forecasting and admonitions to relocate dwelling houses to higher ground were based on severe weather patterns. Then, proactive measures in tsunami engineering for elaborating coastal structures for tsunami defense were sought after the 80 -ft. waves of the 1960 Chilean Tsunami left two million homeless (Pallardy 2017). But in 1983, the unanticipated Japan Sea Earthquake Tsunami wreaked havoc during an otherwise beautiful, fair-weather day. The need for more comprehensive tsunami disaster prevention had become imperative (Shuto and Fujima 2009).

Today, tsunami disaster prevention continues through event forecasting, defense structures, and evacuation. However, tsunami research has evolved into a far more complex subject, and a deeper investigation into tsunami-resistance has begun to develop (Shuto and Fujima 2009). Discussing all the many facets of tsunami research is well beyond the scope of this paper. Instead, the following sections will focus primarily on the methods used for laboratory wave generation. The intent of approaching the literature review in this manner was to determine which methods, if any, could be applicable to the design of the WISD prototype.

## HR Wallingford and University College London, United Kingdom

The collaborative efforts of Hydraulics Research (HR) Wallingford and the University College of London has resulted in the design and construction of three generations of tsunami simulators. The First Generation Tsunami Simulator, developed in 2008, was a 5.9 ft . tall, 3.9 ft . wide, and 15.7 ft . long tank with a variable height outlet placed in a 147.6 ft . long, 3.9 ft . wide, 2-dimensional flume, (see Figure 3). The flume replicated long duration tsunami waves on a 1:50 scale using a pneumatic system and almost 20,000 gallons of water. To generate waves, the First Generation Tsunami Simulator pulled water from the test flume into a sealed tank using a high capacity vacuum pump. Once the tank was full, the vacuum was released using an air control valve. The water rushing back into the test flume took on the desired waveform. The shape of the waveform generated was controlled by adjusting the position of the air control valve. Vacuum pressure in the tank was controlled using a computer automated 45 -degree butterfly valve, (Allsop, Chandler, and Zaccaria 2014).


Figure 3. Simulated model of the First Generation Tsunami Simulator.

In 2011, a CFD model of the First Generation Tsunami Simulator was developed to identify structural modifications that would be necessary to improve upon the initial design. Construction of a Second Generation Tsunami Simulator was undertaken in 2014. The height and width of the tank were increased to 11.5 ft . and 5.9 ft ., respectively. The tank length was decreased to 13.1 ft . and the variable height of the outlet in the First Generation was replaced with a permanent 1.3 ft . outlet. A flow shaper was also added to the outlet to reduce the amount of turbulence generated by the wave, (Allsop, Chandler, and Zaccaria 2014).

The Second Generation Tsunami Simulator flume length and width was likewise increased to 328.1 ft . and 5.9 ft ., respectively. This allowed for more uniform wave development, as well as increased the range of processes and structures to be tested. A second vacuum pump, two ultrasonic level sensors, and updated computer software were also included in the modifications for the new design, (Allsop, Chandler, and Zaccaria 2014).

The Third Generation Tsunami Simulator was developed in 2016 to improve generation of crest-led and trough-led tsunamis. Installed in the HR Wallingford 13.1 ft . wide by 229.7 ft . long Fast Flow Facility, the tank is 13.1 ft . tall, 13.1 ft . wide, and 14.4 ft . long, with an outlet height of 1.31 ft ., (see Figure 4). This latest iteration of tsunami simulator uses a proportional-integral-derivative (PID) feedback system for closed-loop wave generation. Current testing includes the investigation of tsunami run-up, forces and pressures on seawalls, and forces acting on single buildings. (Chandler, Allsop, Granged, and McGovern 2016).


Figure 4. Simulated model of the Third Generation Tsunami Simulator.

All three HR Wallingford Tsunami Simulators use pneumatics in lieu of the traditional piston approach for creating waves. This design allows for longer wave periods to be generated, up to 2 minutes (Lloyd and Rossetto 2012).

## Civil Engineering Department of Clemson University, United States

The rectangular wave tank located at the Flow Physics Laboratory of Clemson University is 40 ft . long, 2 ft wide and 2 ft deep, (see Figure 5). The five modular sections of the wave tank are constructed using clear plexiglass panels, supported by grated steel framing. Waves are generated in the tank using a vertical plate driven by a linear actuator system. The linear actuator system consists of a horizontal actuator, servo-electrical motor, and logic controller. The 5 ft . stroke of the actuator moves the vertical plate in linear proportion to the input voltage signal of the logic controller. The maximum velocity and acceleration achieved by the piston-type system is $5 \mathrm{ft} / \mathrm{s}$ and $32.8 \mathrm{ft} / \mathrm{s}^{2}$, respectively. A sand beach of 1:20 was installed at the opposite end of the tank from the actuator to mimic the environmental conditions of a tsunami-based event (Mohammadi and Testik, 2010).


Figure 5. Piston-type wave generation in the Flow Physics Laboratory wave tank.

## Plymouth University Marine Institute, United Kingdom

The Ocean Basin wave tank at the Coastal Ocean and Sediment Transport (COAST) laboratory of Plymouth University is primarily used for marine renewable energy testing in the UK. Waves up to 3 ft . high are generated by 24 , hydraulically drive, paddle-type wave boards. The basin is approximately 115 ft . long, 50 ft . wide, and 10 ft . deep basin, (see Figure 6). A 6-degree, sloping, moveable floor and recirculating hydraulic system allows the Ocean basin to produce specialty waveforms for both short and long-crest waves (Kirke, Freeman, and Miranda 2015).


Figure 6. Ocean Basin wave tank at Plymouth University, UK.

## LWI Institute for Hydromechanics, The Netherlands

The Deltares Delta Flume is considered one of the largest wave flume facilities in the world at 755 ft long, 16.5 ft wide, and 31 ft high, (see Figure 7). The reservoir for the flume contains 2.4 million gallons of water, approximately the capacity of four Olympic size swimming pools. Three pumping stations are needed to move the water to and from the flume at 264 gallons per second. The Delta Flume uses hydraulic cylinders to move 23 ft . piston-type wave boards. The waves produced by this motion can be up to 15 ft . high with wave periods from 1 to 20 seconds (Streicher, Hofland, and Lindenbergh 2013).


Figure 7. Top view of Delta Deltares wave flume.

## National Institute of Maritime Port and Airport Research, Japan

The Large Hydro-Geo Flume (LHGF) was completed in 2000 under an initiative from the National Institute of Maritime Port and Airport Research Institute (PARI) to promote the redesign of naval and oceanic facilities for tsunami resistance. Large-scale experiments from 1:5 to 1:1 are conducted in the 604 ft . long, 11.5 ft . wide, 40 ft . deep flume to edify the potential failure of coastal structures and foundations, (see Figure 8). The flume also includes a 13 ft . deep bed of sand to simulate sea-seabed interactions (Shimosako, Takahashi, Suzuki, and Kang. 2002).

Regular and irregular waves up to 11.5 ft . are generated with piston-type wavemakers driven by the rack and pinion system of four AC electric servomotors. A current generator is also installed in the flume to produce a $6.5 \mathrm{ft} / \mathrm{s}$ current using two propeller pumps with a rotational speed of 240 RPM (Shimosako, Takahashi, Suzuki, and Kang. 2002).


Figure 8. Large Hydro-Geo Flume at PARI, Japan.

## O.H. Hinsdale Wave Research Laboratory of Oregon State University, United States

The Large Wave Flume at Oregon State University is used for international research on wave-structure interactions, nearshore hydrodynamics and sediment transport, marine renewable energy, tsunami and coastal hazards, and fixed and floating structures. The flume is 342 ft . long, 12 ft . wide, and 15 ft . deep with a maximum water depth of 6.5 ft . for tsunami waves and 9 ft . for wind waves, (see Figure 9). A piston-type, hydraulic actuator moves a horizontal wave board at $13.1 \mathrm{ft} / \mathrm{s}$ using an oil hydraulic pump driven by a 140 hp servo-hydraulic electric motor. The $1: 12$ sloping beach at the end of the flume is used to measure wave runup and tsunami inundation (Rhinefrank, Sschacher, Prudell, and Hammagren 2010).


Figure 9. Large Wave Flume at Oregon State University.

## W. M. Keck Hydraulics Laboratory of California Institute of Technology, United States

The W.H. Keck Hydraulics Laboratory West Tank is 123.8 ft . long, 2 ft . deep, and 15.5 in . wide, (see Figure 10). There are thirteen separate modules that make up the tank. Twelve of the modules are identical, while the thirteenth module includes a movable block section installed by J. L. Hammack in 1972 during his early attempts at tsunami wave generation. The walls of each module were constructed with glass panels, mounted to steel rails (Goring 1979).

The hydraulic system consists of a 40-gallon oil reservoir, a 7.5 hp piston-type pump, wire cloth filters, a 3000 psi unloading valve, two 10 -gallon bladder accumulators, a servo-valve, and two 8 ft . hydraulic cylinders (Goring 1979).


Figure 10. West Tank at W.M. Keck Hydraulics Laboratory

### 3.2 CONCLUSIONS FROM LITERATURE REVIEW

The research conducted for the literature review revealed that the primary mechanism currently in use for wave generation is a hydraulically driven, piston-type wavemaker. The exception was the pneumatic approach used by HR Wallingford and the University College of London. However, the wave characteristics encountered for each of the aforementioned devices were not found to be consistent with the needs of the WISD for a near-vertical wave striking a previously un-inundated area. Furthermore, the space and budgetary requirements of integrating a piston-type or pneumatic method was not practical for the WISD project. The WISD design shall require a new approach to wave generation that has not yet been explored by other organizations.

## 4. DESIGN METHODOLOGY

The design conditions for the WISD were as follows:

1. Prototype shall have a discharge area of approximately $10 \mathrm{ft} . \mathrm{x} 10 \mathrm{ft}$.
2. Scale model shall be $1: 5$ of the prototype ( $2 \mathrm{ft} . \mathrm{x} 2 \mathrm{ft}$.).
3. Wave exiting discharge area shall have a near vertical flow profile.
4. Plates shall be designed for minimal thickness and minimal deflection.
5. Gates shall open near instantaneously.
6. Leakage through closed gates shall be minimal.

The framework for a WISD structural design meeting these design conditions included hand calculations, computational modeling, and finite element analysis. First, hand calculations were performed to determine appropriate plate thicknesses, shaft diameters, and gate requirements for the WISD prototype. These results were then used to create various models in SOLIDWORKS Simulation for finite element analysis (FEA). Once an acceptable design was found to meet all requirements of the project, a $1: 5$ scaled physical model was built for additional testing and refinement for a full-size prototype.

The WISD structural design team was comprised of three CFEL members. Rojin Tuladhar continued the computational FLOW-3D analyses begun by Gregory Roberts in 2015 to identify resultant pressure and configuration requirements for several team concepts. The electrical research and component integration performed by Jash Soumadipta enabled the team to retrieve actual velocity and flow data from the physical model. Larinda Nichols led the design and analysis for all mechanical and structural components to determine feasibility and compliance with the given design conditions.

The work identified in the following thesis is the result of all three members sharing information as they worked concurrently during each stage of the design process. As such, minor design parameters were often adjusted as the investigation into various concepts revealed previously unknown issues or advantages. For each adjustment made, an explanation has been included in this work.

## 5. SOFTWARE VALIDATION

### 5.1 HAND CALCULATIONS

Performing the initial calculations by hand served two purposes. First, the hand calculations provided a general estimate for the thickness requirements of each plate, depending on its support configuration. Second, the results from the hand calculations were then used to validate the SOLIDWORKS computational modeling software.

Due to the large width to thickness ratio anticipated for this application, equations from Roark's Formulas for Stress and Strain ( $6^{\text {th }}$ Edition) were chosen as they specifically address thin metal plates with uniform cross-sections. Case No. 6 of Roark's Table 26, "Formulas for flat plates with straight boundaries and constant thickness" for a "rectangular plate, two long edges fixed, two short edges simply supported" was used under the assumption that each plate will be simply supported length-wise by the WISD housing using clamps or bolts. This case is illustrated below in Figure 11.


Figure 11. Case No. 6 for flat plates with straight boundaries and constant thickness (Young 1989).

Based on Case No. 6, the maximum stress at the center of the long edge of each plate can be found using Equation 1 below, where $q$ represents a uniform pressure to the plate (in psi), $b$ represents the width (in inches), $a$ represents length (in inches) and $t$ is the plate thickness (in inches). Specific variables for $\beta$ are tabulated for each length to width ratio, (see Table 1).

$$
\begin{equation*}
\operatorname{Max} \sigma=\frac{-\beta q b^{2}}{t^{2}} \tag{1}
\end{equation*}
$$

Table 1. Tabulated Specific Values for $\beta$ (Young 1989)

| $\mathbf{a} / \mathbf{b}$ | $\mathbf{1}$ | $\mathbf{1 . 2}$ | $\mathbf{1 . 4}$ | $\mathbf{1 . 6}$ | $\mathbf{1 . 8}$ | $\mathbf{2}$ | $\infty$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $\boldsymbol{\beta}$ | 0.4182 | 0.4626 | 0.4860 | 0.4968 | 0.4971 | 0.4973 | 0.500 |

The maximum deflection at the center of the plate can be found using Equation 2, where $q, b$, and $t$ are the same values used in Equation 1, and $E$ represents the Modulus of Elasticity of the material. Tabulated specific values for $\alpha$ are shown in Table 2.

$$
\begin{equation*}
\operatorname{Max} y=\frac{-\alpha q b^{4}}{E t^{3}} \tag{2}
\end{equation*}
$$

## Table 2. Tabulated Values for $\alpha$ (Young 1989)

| $\mathbf{a} / \mathbf{b}$ | $\mathbf{1}$ | $\mathbf{1 . 2}$ | $\mathbf{1 . 4}$ | $\mathbf{1 . 6}$ | $\mathbf{1 . 8}$ | $\mathbf{2}$ | $\infty$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $\boldsymbol{\alpha}$ | 0.0210 | 0.0243 | 0.0262 | 0.0273 | 0.0280 | 0.0283 | 0.0285 |

The preliminary design conditions specified that each plate must support approximately 1 ft . of water. The hydrostatic pressure at this depth was found using the equation for hydrostatic pressure.

$$
\begin{equation*}
P=\gamma h \tag{3}
\end{equation*}
$$

Assuming general atmospheric conditions at sea level, the specific weight of water $(\gamma)$ is $63.4 \mathrm{lb} / \mathrm{ft}^{3}$, or approximately $0.0361 \mathrm{lb} / \mathrm{in}^{3}$. Multiplying the specific weight by the 12 -inch depth, the pressure of the water was found to be 0.4333 psi . In Roark's formula's, pressure is represented by the variable q in the equations above. Also, the material properties used for the A36 carbon steel plates are shown in Table 3.

Table 3. Material Properties of A36 Plain Carbon Steel (Hibbeler 2005)

| Elastic Modulus (psi) | Poisson's Ratio | Yield Strength (psi) | Mass Density (lb/in³) |
| :---: | :---: | :---: | :---: |
| $30,457,925$ | 0.3 | 31,995 | 0.282 |

Equations 1 and 2 were used along with values obtained from Table 1, Table 2, and Table 3 to determine the maximum stress and deflection of $10 \mathrm{ft} . \times 10 \mathrm{ft}$., $5 \mathrm{ft} . \times 10 \mathrm{ft}$., and 2-1/2 ft. x 10 ft . plates of various thickness when supporting 1 ft . of water. Details of these calculations can be found in Appendix A, p 78.

Table 4. Maximum Yield and Deflection for a 10 ft. x 10 ft. Plate Using Roark's Formulas

|  | Ratio <br> $(\mathbf{w / l})$ | Tabulated <br> Value | Tabulated <br> Value | Pressure <br> $(\mathbf{p s i})$ | Weight <br> $(\mathbf{l b})$ | Max Yield <br> $(\mathbf{p s i})$ | Max Deflect <br> $(\mathbf{i n})$ |
| :--- | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | $\mathbf{a / b}$ | $\boldsymbol{\beta}$ | $\boldsymbol{\alpha}$ | $\mathbf{q}$ | $\mathbf{w}$ | $\boldsymbol{\sigma}$ | $\boldsymbol{y}$ |
| $\mathbf{3 / 8} \mathbf{~ i n . ~}$ | 1 | 0.4182 | 0.021 | 0.433 | 1522 | 18557 | 1.175 |
| $\mathbf{1 / 4 ~ i n . ~}$ | 1 | 0.4182 | 0.021 | 0.433 | 1015 | 41753 | 3.965 |
| $\mathbf{3 / 1 6}$ in. | 1 | 0.4182 | 0.021 | 0.433 | 761 | 74228 | 9.399 |
| $\mathbf{1 / 8} \mathbf{~ i n . ~}$ | 1 | 0.4182 | 0.021 | 0.433 | 507 | 167012 | 31.72 |

Table 5. Maximum Yield and Deflection for a 5 ft. x 10 ft. Plate Using Roark's Formulas

|  | Ratio <br> $(\mathbf{w} / \mathbf{)}$ | Tabulated <br> Value | Tabulated <br> Value | Pressure <br> $(\mathbf{p s i})$ | Weight <br> $(\mathbf{( b )})$ | Max Yield <br> $(\mathbf{p s i})$ | Max Deflect <br> $(\mathbf{i n})$ |
| :--- | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | $\mathbf{a / b}$ | $\boldsymbol{\beta}$ | $\boldsymbol{\alpha}$ | $\mathbf{q}$ | $\mathbf{w}$ | $\boldsymbol{\sigma}$ | $\boldsymbol{y}$ |
| $\mathbf{3 / 8} \mathbf{i n .}$ | 2 | 0.4973 | 0.0283 | 0.433 | 761 | 5517 | 0.099 |
| $\mathbf{1 / 4} \mathbf{i n}$. | 2 | 0.4973 | 0.0283 | 0.433 | 507 | 12413 | 0.334 |
| $\mathbf{3 / 1 6}$ in. | 2 | 0.4973 | 0.0283 | 0.433 | 380 | 22067 | 0.792 |
| $\mathbf{1 / 8}$ in. | 2 | 0.4973 | 0.0283 | 0.433 | 254 | 49650 | 2.672 |

Table 6. Maximum Yield and Deflection for a 2-1/2 ft. x 10 ft. Plate Using Roark's Formulas

|  | Ratio <br> $(\mathbf{w} / \mathbf{l})$ | Tabulated <br> Value | Tabulated <br> Value | Pressure <br> $(\mathbf{p s i})$ | Weight <br> $(\mathbf{l b})$ | Max Yield <br> $(\mathbf{p s i})$ | Max Deflect <br> $(\mathbf{i n})$ |
| :--- | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | $\mathbf{a / b}$ | $\boldsymbol{\beta}$ | $\boldsymbol{\alpha}$ | $\mathbf{q}$ | $\mathbf{w}$ | $\boldsymbol{\sigma}$ | $\mathbf{y}$ |
| $\mathbf{3 / 8} \mathbf{i n .}$ | 4 | 0.5000 | 0.0285 | 0.433 | 380 | 1387 | 0.006 |
| $\mathbf{1 / 4} \mathbf{~ i n . ~}$ | 4 | 0.5000 | 0.0285 | 0.433 | 254 | 3120 | 0.021 |
| $\mathbf{3 / 1 6} \mathbf{~ i n . ~}$ | 4 | 0.5000 | 0.0285 | 0.433 | 190 | 5547 | 0.050 |
| $\mathbf{1 / 8} \mathbf{~ i n . ~}$ | 4 | 0.5000 | 0.0285 | 0.433 | 127 | 12480 | 0.168 |

Here it should be noted that for these formulas to remain applicable to the WISD plates, the following assumptions from Roark's Formulas for Stress and Strain must hold true: "(1) the plate is flat, of uniform thickness, and of homogeneous isotropic material; (2) the thickness is not more than about onequarter of the least transverse dimension, and the maximum deflection is not more than about one-half the thickness; (3) all forces - loads and reactions - are normal to the plane of the plate; (4) the plate is nowhere stressed beyond the elastic limit" (Young 1989).

All results from Table 4, Table 5, and Table 6 were in agreement with assumptions (1) and (3). However, assumption (4) requires that the maximum stress does not exceed the yield stress of the material. In comparing each result to the maximum yield strength of the A36 steel shown in Table 3, only the eight plates shown in Table 7 met assumption (4).

Table 7. Plate Geometries Meeting Roark's Requirement (4)

| Plate Size | $\begin{gathered} \hline \text { Weight (lb) } \\ w \\ \hline \end{gathered}$ | $\begin{gathered} \text { Max Yield (psi) } \\ \sigma \\ \hline \end{gathered}$ |
| :---: | :---: | :---: |
| $10 \mathrm{ft} . \times 10 \mathrm{ft} . \times 3 / 8 \mathrm{in}$. | 1522 | 18557 |
| 5 ft . 10 ft . $\mathrm{x} 3 / 8 \mathrm{in}$. | 761 | 5517 |
| 5 ft . 10 ft . $\times 1 / 4 \mathrm{in}$. | 507 | 12413 |
| 5 ft . x 10 ft . $\mathrm{x} 3 / 16 \mathrm{in}$. | 380 | 22067 |
| 2-1/2 ft. $\times 10 \mathrm{ft}$. $\times 3 / 8 \mathrm{in}$. | 380 | 1387 |
| 2-1/2 ft. x $10 \mathrm{ft} . \times 1 / 4 \mathrm{in}$. | 254 | 3120 |
| 2-1/2 ft. x 10 ft . $\mathrm{x} 3 / 16 \mathrm{in}$. | 190 | 5547 |
| 2-1/2 ft. x 10 ft . $\times 1 / 8 \mathrm{in}$. | 127 | 12480 |

The final assumption, (3), requires a plate deflection less than half of the original thickness. Therefore, the deflection of each plate in Table 7 was then compared to its respective thickness. The four plates listed below in Table 8 were the only configurations shown to meet all four requirements.

Table 8. Plate Geometries Meeting All Four Roark's Requirements

| Plate Size | Weight <br> (lb) | $\underset{(\mathbf{p s i})}{\text { Max Yield }}$ | Factor of Safety | Max Deflect (in) | Max Deflect per Plate Thickness |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  | w | $\sigma$ | N | y |  |
| 5 ft . 10 ft . $\mathrm{x} 3 / 8 \mathrm{in}$. | 761 | 5517 | 5.8 | 0.099 | 0.26 |
| 2-1/2 ft. x $10 \mathrm{ft} . \times 3 / 8 \mathrm{in}$. | 380 | 1387 | 23 | 0.006 | 0.02 |
| 2-1/2 ft. x $10 \mathrm{ft} . \times 1 / 4 \mathrm{in}$. | 254 | 3120 | 10 | 0.021 | 0.08 |
| 2-1/2 ft. x 10 ft . $\times 3 / 16 \mathrm{in}$. | 190 | 5547 | 5.8 | 0.050 | 0.27 |

After establishing reasonable solutions for plate thickness using the hand calculations above, each of the four configurations of Table 8 were modeled and using SOLIDWORKS computational software. All material properties, fixtures, and external load (pressure) were simulated to replicate the parameters and conditions assumed in the hand calculations. The FEA analysis for each study took approximately twenty minutes using SOLIDWORKS 2016 x 84 Student Edition on an Intel(R)Core(M) i7-7700HQ CPU @ 2.80 GHz and 16.0 GB RAM. Mesh settings were set to solid, fine, and curvature-based. The results for each case are discussed below. Additional details of SOLIDWORKS Simulation settings can be found in Appendix B, p 89.

### 5.2 COMPARISON OF SOLIDWORKS RESULTS TO HAND CALCULATIONS

Validation 1: 5 ft. x 10 ft. x 3/8 in. A36 Steel Plate
The FEA analysis of the $5 \mathrm{ft} . \times 10 \mathrm{ft} . \mathrm{x} 3 / 8 \mathrm{in}$. plate showed a maximum yield stress of 5109 psi , approximately 400 psi less than the maximum yield stress calculated using Roark's formulas, (see Figure 12). The FEA maximum deflection was 0.111 in ., a slight increase from the hand calculated value of 0.099 in, (see Figure 13).


Figure 12. Von-Mises stress of the 5 ft. $x 10$ ft. $x 3 / 8$ in. plate with long edges fixed under 0.433 psi.


Figure 13. Deflection of the 5 ft. $x 10$ ft. $x 3 / 8$ in. plate with long edges fixed under 0.433 psi.

## Validation 2: 2-1/2 ft. x 10 ft. x 3/8 in. A36 Steel Plate

The FEA analysis of the $2-1 / 2 \mathrm{ft}$. $\times 10 \mathrm{ft} . \times 3 / 8 \mathrm{in}$. plate showed a maximum yield stress of 1191 psi , approximately 200 psi less than the maximum yield stress calculated using Roark's formulas, (see Figure 14). The FEA maximum deflection was 0.007 in ., in close agreement with the hand calculated value of 0.006 in., (see Figure 15).


Figure 14. Von-Mises stress of 2-1/2 ft. x 10 ft. x $3 / 8$ in. plate with long edges fixed under 0.433 psi.


Figure 15. Deflection of 2-1/2 ft. x 10 ft. x $3 / 8$ in. plate with long edges fixed under 0.433 psi.

## Validation 3: 2-1/2 ft. x $10 \mathrm{ft} . \times 1 / 4 \mathrm{in}$. A36 Steel Plate

The FEA analysis of the $2-1 / 2 \mathrm{ft}$. x 10 ft . x $1 / 4 \mathrm{in}$. plate showed a maximum yield stress of 2689 psi , approximately 400 psi less than the maximum yield stress calculated using Roark's formulas, (see Figure 16). The FEA maximum deflection was 0.020 in ., a slight increase from the hand calculated value of 0.021 in., (see Figure 17).


Figure 16. Von-Mises stress of 2-1/2 ft. x 10 ft. $x$ 1/4 in. plate with long edges fixed under 0.433 psi.


Figure 17. Deflection of 2-1/2 ft. x 10 ft. x 1/4 in. plate with long edges fixed under 0.433 psi.

## Validation 4: 2-1/2 ft. $\mathbf{x} 10$ ft. $\times \mathbf{3 / 1 6}$ in. A36 Steel Plate

The FEA analysis of the $2-1 / 2 \mathrm{ft}$. x 10 ft . x $3 / 16 \mathrm{in}$. plate showed a maximum yield stress of 4703 psi , approximately 850 psi less than the maximum yield stress calculated using Roark's formulas, (see Figure 18). The FEA maximum deflection was 0.055 in., a slight increase from the hand calculated value of 0.050 in ., (see Figure 19).


Figure 18. Von-Mises stress of 2-1/2 ft. x 10 ft. x $3 / 16$ in. plate with long edges fixed under 0.433 psi.


Figure 19. Deflection of 2-1/2 ft. x 10 ft. $x$ 3/16 in. plate with long edges fixed under 0.433 psi.

Table 9 compares the results obtained using the SOLIDWORKS FEA versus those found using Roark's Formulas. Overall percent difference was within $16.5 \%$ for yield stress and $11.5 \%$ for deflection.

Table 9. Comparison of Hand Calculation and SOLIDWORKS FEA Results

| PLATE SIZE | MAXIMUM YIELD STRESS (psi) |  |  | MAXIMUM DEFLECTION (in) |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | ROARK'S | SOLIDWORKS | \% DIFF | ROARK'S | SOLIDWORKS | \% DIFF |
| 5 ft . x $10 \mathrm{ft} . \times 3 / 8 \mathrm{in}$. | 5517 | 5109 | 7.70 | 0.099 | 0.111 | 11.5 |
| 2-1/2 ft. x 10 ft . $\mathrm{x} 3 / 8 \mathrm{in}$. | 1387 | 1191 | 15.2 | 0.006 | 0.007 | 10.9 |
| 2-1/2 ft. x 10 ft . $\times 1 / 4 \mathrm{in}$. | 3120 | 2689 | 14.9 | 0.021 | 0.023 | 10.5 |
| 2-1/2 ft. x 10 ft . $\mathrm{x} / 16 \mathrm{in}$. | 5547 | 4703 | 16.5 | 0.050 | 0.055 | 9.30 |

The uncertainty associated with Roark's Formulas is given by Young as approximately 7\% due to the experimentally derived values for $\alpha$ and $\beta$ used in each equation. SOLIDWORKS does not explicitly state uncertainty values for FEA results since each result is heavily dependent on the mesh settings used. Finer mesh settings require more complex calculations to be performed by the program; thereby, increasing the reliability of each outcome.

Both methods were deemed in satisfactory agreement, and the software was considered reliable for all further design. Validation was important since the model would become more complex as additional elements were added such as shafts, rods, gates, multiple plate configurations, etc., and hand calculations would be difficult and highly susceptible to more errors.

## 6. WISD PROTOTYPE DESIGN

Establishing an acceptable structural design for WISD was an iterative process based on preliminary hand calculations and computational modelling. First, adequate plate thickness had to be identified before an accompanying gate system could be developed. Then, the configuration of the gate system would dictate the mechanical aspects of WISD operation. Finally, a comprehensive review of the final design was necessary to ensure all components would operate as designed and not interfere with the processes of one another.

### 6.1 DESIGN AND ANALYSIS OF WISD PLATES

Establishing a plate thickness, geometry, and configuration best suited for the WISD prototype was the first step in the design process. The results of the software validation indicated that thinner plates could be used if the spanning width was decreased. Plates spanning the entire $10-\mathrm{ft}$. width would require additional thickness, but they would be less invasive on the developing wave than multiple sections per channel. Design iterations were used to identify the advantages and disadvantages of each option.

## Plate Design Iteration 1: Four 2-1/2 ft. Plates per 10 ft. Wide Channel

Model Details: Four 2.5 ft x 10 ft x $3 / 32 \mathrm{in}$. A36 steel plates were designed to span the $10-\mathrm{ft}$. wide channel. A single plate was modeled with the long edges fixed from translation and rotation. A total of 36 plates would be required for the entire prototype, (see Figure 20). A uniform pressure of 0.433 psi was applied over the face of the plate.

FEA Results: The maximum von-Mises stress was observed to be 17,446 psi, (see Figure 21), with a maximum deflection of 0.044 in. , (see Figure 22).

Remarks: The maximum stress did not exceed the 36,000 psi yield stress of A36 steel, and the 0.044 in. deflection was considered minimal. However, it was unclear how dividing the 10 ft . channel into four separate $2-1 / 2 \mathrm{ft}$. sections might affect the target flow profile. FLOW-3D models of the multiple channel configuration were created by WISD member, Rojin Tuladhar. The vertical flow profile appeared unchanged from a single channel of 10 -ft width but installing supports for each section was considered a disadvantage. The team agreed more options should be explored.


Figure 20. Overview of the 2-1/2 ft. x 10 ft. $x$ 3/32 in. plate system.


Figure 21. Von-Mises stress of a 2-1/2 ft. x 10 ft. $x$ 3/32 in. plate with long edges fixed under 0.433 psi.


Figure 22. Deflection of a 2-1/2 ft. x 10 ft. x 3/32 in. plate with long edges fixed under 0.433 psi.

Plate Design Iteration 2: A Single 10 ft . Wide Plate per 10 ft . Channel
Model Details: A 10 ft x 10 ft . x 1 in. thick plate was modeled with long edges fixed. A total of 9 plates would be required for the reservoir section of the prototype, (see Figure 23). A uniform pressure of 0.433 psi was applied over the face of the plate.

FEA Results: Maximum von-Mises stress was 2,743 psi, (see Figure 24). Maximum deflection was 0.099 in., (see Figure 25).

Remarks:
Each 10 ft . x 10 ft . x 1 in . plate would weigh approximately $4,084 \mathrm{lbs}$., a significant issue for installation and safety issues. A 1 in. thick plate was also scrutinized as being highly disruptive to a developing flow profile. The solution was to use a thinner plate and investigate different options for supporting it.


Figure 23. Overview of the 10 ft. x 10 ft. $x 1$ in. plate system.


Figure 24. Von-Mises stress of a 10 ft. x 10 ft. $x 1$ in. plate with long edges fixed under 0.433 psi.


Figure 25. Deflection stress of a 10 ft. $x 10$ ft. $x 1$ in. plate with long edges fixed under 0.433 psi.

Plate Design Iteration 3: Plate Thickness Decreased and Number of Fixed Edges Increased
Model Details: A 10 ft x 10 ft x $3 / 8 \mathrm{in}$. plate was modeled with three sides fixed in place. A uniform pressure of 0.433 psi was applied over the face of the plate.

FEA Results: Maximum von-Mises stress was 19,327 psi, (see Figure 26). Maximum deflection was 1.852 in., (see Figure 27).

Remarks: The 19,327 psi maximum stress did not exceed the 36,000 psi yield stress of A36 steel, but the deflection was substantial at 1.852 in.. Additional support was deemed necessary.


Figure 26. Von-Mises stress of a 10 ft. x 10 ft. $x 3 / 8$ in. plate with three fixed edges under 0.433 psi.


Figure 27. Deflection of a 10 ft. $x 10$ ft. $x$ 3/8 in. plate with three fixed edges under 0.433 psi.

Plate Design Iteration 4: Fixed Edges Increased to all Four Sides of Plate
Model Details: A 10 ft . x 10 ft . x 3/8 in. thick plate was modeled with all four sides fixed. A uniform pressure of 0.433 psi was applied over the face of the plate.

FEA Results: Maximum von-Mises stress was 18,098 psi, (see Figure 28). Maximum deflection was 1.630 in., (see Figure 29).

Remarks: The maximum stress was reduced from 19,327 to $18,098 \mathrm{psi}$, and the deflection was reduced from 1.852 in. to 1.630 in . The deflection was still considered excessive. The team began research on thin supports to decrease deflection at the center of the plate.


Figure 28. Von-Mises stress of a 10 ft. $x 10$ ft. $x 3 / 8$ in. plate with three fixed edges under 0.433 psi.


Figure 29. Deflection of a 10 ft. $x 10$ ft. $x 1$ in. plate with three fixed edges under 0.433 psi.

It was at this time in the design process that the concurrent FLOW-3D research being performed by WISD member Rojin Tuladhar showed that increasing the hydrostatic head from 1 ft to 3 ft . would aid in the development of a near vertical flow profile. The FLOW-3D simulation also indicated as much as 20 psi of air pressure would be required to accelerate the water from stagnation to a constant $25.4 \mathrm{ft} / \mathrm{s}$ velocity before it exists the conduit.

The change in design parameters from 0.433 psi to approximately 21.3 psi substantially increased the difficulty in developing an adequately thin, yet adequately strong prototype plate. The need for additional plate supports became imperative. It was assumed that adding supports to the reservoir section would have less of an impact on the target velocity profile than adding supports to the plates located downstream of the gate system. Furthermore, the water contained in the post-gate section would not be static. It would be in constant motion when acting upon by the 20 psi horizontal, pneumatic force. As such, the design focus was shifted from the reservoir plates to the post-gate plates to determine the stress and deflection behavior under these new parameters.

## Plate Design Iteration 5: Investigation of Static vs. Dynamic Effects on Post Gate Plates

Model Details: Nine 10 ft x 5 ft x $1 / 2 \mathrm{in}$. plates were modeled with three edges fixed in place (exit edge remained simply supported). All nine plates were included in the simulation to observe the overall behavior of the plate system, (see Figure 30). Two FEA studies were performed to compare the behavior of all nine plates when subjected to a static, then dynamic load. For the static study, a uniform 1.3 psi pressure was applied vertically to the top plane of each plate. The dynamic study included a non-linear, time-dependent study based on the 20 psi horizontal force also acting on the water column.

FEA Results: The static analysis showed a maximum von-Mises stress of 8,864 psi would develop in all nine plates when a 1.3 psi pressure is applied, (see Figure 31). The maximum
deflection was 0.413 in., (see Figure 32). In the time-dependent analysis, the maximum von-Mises stress was 8,372 psi, (see Figure 33), and the maximum deflection was 0.442 in. for the 20 psi horizontal and 1.3 psi vertical loads, (see Figure 34 and Figure 35).

Remarks: The time-dependent simulation used 67 time steps from the onset of deflection to cessation. Figure 34 and Figure 35 each contain six time step images showing the deflection at each respective time step. The stress to each plate at maximum deflection is shown in Figure 33. A comparison between the two studies confirmed flowing water through all nine plates would yield maximum stress and deflection values similar to those of the static study. The minor variance between the two stress and deflection values were instead attributed to using two different FEA methods (static linear vs. dynamic nonlinear) to solve the system. Regardless, the nearly $1 / 2 \mathrm{in}$. deflection was not considered acceptable. The plate thickness could not be increased any further, and the investigation into an alternative support system continued.


Figure 30. Overview of the 10 ft. $x 5$ ft. $x$ 1/2 in. post-gate system.


Figure 31. Von-Mises stress for nine 10 ft. $x 5$ ft. $x$ 1/2 in. plates under a 1.3 psi static load.


Figure 32. Deflection for the nine 10 ft. $x 5$ ft. $x$ 1/2 in. plates under a 1.3 psi static load.


Figure 33. Von-Mises stress for the nine 10 ft. $x 5$ ft. $x 1 / 2$ in. plates under static and dynamic loading.

$t \approx 0.01 \mathrm{~s}$, max deflection $\approx 0.053 \mathrm{in}$.

$t \approx 0.10$ s, max deflection $\approx 0.278 \mathrm{in}$.

$t \approx 0.2 s$, max deflection $=0.442 \mathrm{in}$.

$t \approx 0.05 \mathrm{~s}$, max deflection $\approx 0.201 \mathrm{in}$.

$t \approx 0.15 \mathrm{~s}$, max deflection $\approx 0.370 \mathrm{in}$.


Deflection of 0.442 in. from $t \approx 0.2 \mathrm{~s}$ to $t \approx 0.6 \mathrm{~s}$

Figure 34. Onset of deflection of the 10 ft. $x 5$ ft. $x 1 / 2$ in. plates under static and dynamic loading.

$$
(t=0.01 \text { to } t=0.2 \text { seconds })
$$


$t \approx 0.59 \mathrm{~s}$, max deflection $=0.442 \mathrm{in}$.

$t \approx 0.71 \mathrm{~s}$, max deflection $\approx 0.201 \mathrm{in}$.

$t \approx 0.80 \mathrm{~s}$, max deflection $=0.0 \mathrm{in}$.

$t \approx 0.65 s$, max deflection $\approx 0.334 \mathrm{in}$.

$t \approx 0.75 \mathrm{~s}$, max deflection $\approx 0.103 \mathrm{in}$.


No deflection after $t=0.8 \mathrm{~s}$, plates unloaded.

Figure 35. Cessation of deflection of the $10 \mathrm{ft} x$.5 ft. $x$ 1/2 in. plates under static and dynamic loading.

$$
(t=0.6 \text { to } t=0.9 \text { seconds })
$$

## Plate Design Iteration 6: Cross Beams Added to Plate for Additional Support

Model Details: A $10 \mathrm{ft} . \mathrm{x} 5 \mathrm{ft} . \mathrm{x} 3 / 8 \mathrm{in}$. plate was modeled with long edges fixed. Cross beams measuring $1 \mathrm{in} . \mathrm{x} 1 / 4 \mathrm{in}$. x 10 ft . were attached to the underside of the plate at 4 ft . intervals, (see Figure 36). A uniform pressure of 1.3 psi was applied over the face of the plate, accounting only for hydrostatic pressure.

FEA Results: Maximum von-Mises stress was observed to be $4,873 \mathrm{psi}$, (see Figure 37), with a maximum deflection of 0.101 in., (see Figure 38).

Remarks: Both maximum stress and deflection were considered acceptable; however, the interference of a horizontal support system with the target flow profile was unknown. Thin vertical supports would be a less invasive option.


Figure 36. Bottom view of crossbeams attached to a 10 ft. $x 5$ ft. $x$ 3/8 in. plate.


Figure 37. Von-Mises stress of a 10 ft. $x 5$ ft. $x$ 3/8 in. plate with crossbeams under 1.3 psi.


Figure 38. Deflection of a 10 ft. x 5 ft. $x$ 3/8 in. plate with crossbeams under 1.3 psi.

Plate Design Iteration 7: A Thin Steel Support Installed at Exit Edge of Post-Gate Plates
Model Details: A $10 \mathrm{ft} . \mathrm{x} 5 \mathrm{ft}$. x $3 / 8 \mathrm{in}$. plate was modeled with long edges fixed in place. A $1 / 4 \mathrm{in}$. thick, steel vertical support was fixed to the plate using $3 / 8 \mathrm{in}$. notches at 12 in . intervals, (see Figure 39). A uniform pressure of 1.3 psi was applied over the face of the plate.

FEA Results: Maximum von-Mises stress was 17,074 psi, (see Figure 40), and maximum deflection was 0.383 in., (see Figure 41).

Remarks: The appearance of the stress concentration surrounding the vertical support was a concern for future cracking and/or failure in the support. But the overall improvement of placing a vertical support in the center of each plate was considered progress.


Figure 39. A 10 ft. $x 5$ ft. $x$ 3/8 in. plate supported using 1/4 in. thick, notched steel.


Figure 40. Von-Mises stress of a 10 ft. $x 5$ ft. $x 3 / 8$ in. plate under 1.3 psi with thin vertical support.


Figure 41. Deflection of a 10 ft. $x 5$ ft. $x$ 3/8 in. plate under 1.3 psi with thin vertical support.

## Plate Design Iteration 8: Thin Vertical Support Replaced by Two Rods

Model Details: A 10 ft . x $5 \mathrm{ft} . \times 3 / 8 \mathrm{in}$. plate was modeled with long edges fixed in place. The vertical support of Plate Design Iteration 8 was replaced by two $1 / 4 \mathrm{in}$. steel rods, fixed in tension to the top and bottom of the device housing, (see Figure 42). Washers were attached to at each opening in the plate to provide reinforcement, (see Figure 43). The rod was modeled as fixed to each washer and rod under the assumption that the components would be welded in the physical system. The distance from the edge of the plate to the first and second rod was 6 in . and 18 in ., respectively. A uniform pressure of 1.3 psi was applied over the face of the plate.

FEA Results: The maximum von-Mises stress was 16,254 psi, (see Figure 44), and the maximum deflection was 0.049 in., (see Figure 45).

Remarks: High stress concentration were observed at the location of each fixture, even with the addition of the reinforcing washers. The deflection, however, was significantly decreased by 0.334 in.. It was hypothesized that additional rods would mitigate the stress concentrations but welding several threaded rods to each limited access plate was not feasible.


Figure 42. Overview of vertical rod plate support.


Figure 43. Close-up of vertical rod and washer attachments.


Figure 44. Von-Mises stress of a $10 \mathrm{ft} . \mathrm{x} 5 \mathrm{ft} . x 3 / 8$ in. plate under 1.3 psi with two support rods.


Figure 45. Deflection of a 10 ft. $x 5$ ft. $x$ 3/8 in. plate under 1.3 psi with two support rods.

## Plate Design Iteration 9: Threaded Rods and Rivet Nuts for Additional Plate Support

Model Details: One 10 ft x 5 ft x $3 / 8 \mathrm{in}$. was modeled with long edges fixed in place. Threaded rivet nuts were installed in intervals across each plate, (see Figure 46). Six $1 / 4 \mathrm{in}$. steel threaded rods are threaded through each rivet nut, then fixed in tension to the top and bottom of the device housing, (see Figure 47). The distance from the edge to the first and second rod in the center of the plate was 12 in . and 24 in ., respectively. The distance from the edge to the first and second rod on each side of the center supports was 6 in . and 18 in., respectively. A uniform pressure of 1.3 psi was applied over the face of the plate.

FEA Results: The maximum von-Mises stress was $15,801 \mathrm{psi}$, (see Figure 48), and maximum deflection was 0.043 in., (see Figure 49).

Remarks: The 15,801 psi maximum von-Mises stress in the plate was reduced to almost half the 36,000 psi yield stress of A36 steel. The 0.043 in. maximum deflection was small enough
to be considered reasonably negligible. The concept of using thin, threaded rods coupled with low-profile rivet nuts had succeeded in distributing the stress and deflection of each plate more uniformly than had previous design iteration. Thin vertical rods also meant less disruption to the wave development. The system was designed for each rod to be threaded continuously through each rivet nut in the plates below from above the WISD housing. This provided flexibility for future modifications since adding or removing supports could be done with ease as necessary. More information about the large flange rivet nuts used for this design can be found in Appendix C, p 90.


Figure 46. Close-up of expanded rivet nut after installation.
(Plate, rod, and threads removed for clarity.)


Figure 47. Overview of threaded rivet nut and rod plate supporting system.


Figure 48. Von-Mises stress of a 10 ft. $x 5$ ft. $x 3 / 8$ in. plate under 1.3 psi with rivet nut/rod supports.


Figure 49. Deflection of a 10 ft. $x 5$ ft. $x 3 / 8$ in. plate under 1.3 psi with rivet nut/rod supports.

## Summary of Plate Design Iterations

The results of the nine plate design iterations are shown below in Table 10.

Table 10. Summary of plate design iteration results

| ITER. | WIDTH | LENGTH | THICK. | SUPPORT | PRESSURE | STRESS | DEFLECT. |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | $2-1 / 2 \mathrm{ft}$. | $10 \mathrm{ft} .$ | 3/32 in. | Long sides fixed | 0.433 psi | 17,446 psi | $0.044 \mathrm{in} .$ |
| 2 | 10 ft . | 10 ft . | $1 \mathrm{in} .$ | Long sides fixed | 0.433 psi | 2,743 psi | 0.099 in. |
| 3 | 10 ft . | 10 ft . | $3 / 8 \mathrm{in}$. | Three sides fixed | 0.433 psi | 19,327 psi | 1.852 in . |
| 4 | 10 ft . | 10 ft . | 3/8 in. | All sides fixed | 0.433 psi | 18,098 psi | 1.630 in . |
| 5 | 5 ft . | 10 ft . | $1 / 2 \mathrm{in}$. | Three sides fixed | 1.3 psi | 8,864 psi | 0.413 in. |
| 6 | 5 ft . | 10 ft . | $3 / 8 \mathrm{in}$. | Crossbeams | 1.3 psi | 4,873 psi | 0.101 in. |
| 7 | 5 ft . | 10 ft . | $3 / 8 \mathrm{in}$. | Thin vertical sheet | 1.3 psi | 17,074 psi | 0.383 in. |
| 8 | 5 ft . | 10 ft . | $3 / 8 \mathrm{in}$. | Two welded rods | 1.3 psi | 16,254 psi | 0.049 in. |
| 9 | 5 ft . | 10 ft . | 3/8 in. | Six threaded rods | 1.3 psi | 15,801 psi | 0.043 in. |

### 6.2 DESIGN AND ANALYSIS OF WISD GATES

The success of the WISD is largely dependent on the functionality of the gate system. As part of the project requirements, the gate system must have the following capabilities:

1. Gates must be designed to withstand high water pressures before water is released.
2. Gate design should not interfere with the target flow profile.
3. All gates must open near instantaneously.
4. Leakage from reservoir prior to gates opening should be kept at a bare minimum.

The FLOW-3D simulations performed by Greg Roberts, used the concept of ten bottom hinged-gates, opening each channel simultaneously as the water reservoir behind them became pressurized (Roberts 2016). As such, each gate would be a rectangular valve driven by a single, rotating shaft. A gate thickness and shaft size capable of withstanding both hydrostatic and pneumatic pressures from the reservoir section would need to be determined.

The amount of torque on the gate shaft was calculated under two main assumptions. First, the shaft was assumed fixed on each end (although the mechanism for holding the shaft in place had not yet been determined). And second, the gate would be permanently fixed to the shaft, and both components would be potentially subjected to 21.3 psi of hydrostatic and pneumatic pressure. Each gate must span the entire 10 ft . width of the channel; therefore, the resultant deflection and stress from the applied pressure would ultimately determine the configuration best suited for the task.

The method for attaching the gate to the shaft also presented the team with an obstacle. Since the shaft would need to drive the movement of the gate, the two required a strong, permanent bond. A keyed shaft is a standard rod designed with a machined groove running the length of the shaft to accommodate the placement of a key. The key is a small piece of metal fabricated to fit snugly into the groove of the shaft and accommodate other keyed components, such as gears and pulleys. The attached components can then
be driven by the rotation of the shaft. Using the same concept, a thin plate of metal with a thickness equivalent to the respective key was assumed fixed into the keyway of the shaft using an all-around filet weld at the interface between shaft and plate. The shaft was designed to extend past the plate for 3 in . in each direction to allow space for the driving mechanism to be later installed. An overview of the simple gate created by the keyed shaft and plate is shown below in Figure 50.


Figure 50. Isometric view of gate plate attached to keyed shaft.
(Left: keyed shaft, Center: plate inserted into keyway, Right: all around fillet weld)

The 10 ft . channel span of the shaft, coupled with the 21.3 psi pressure acting on the gate from the water and compressed air would cause a large bending moment to occur on both pieces. To determine the extent of the bending moment, hand calculations from Shigleys Mechanical Engineering Design were used, (see Appendix D, p 92). The first calculation was performed to determine the minimum diameter required to withstand the 20 psi pneumatic pressure and 1.3 psi hydrostatic pressure using the full 36,000 psi yield strength of A36 steel. The resulting diameter was 3.88 in . The next calculation removed the 20 psi pneumatic pressure to observe a minimum diameter requirement for only the 1.3 psi hydrostatic pressure, assuming the gates and compressed air could be released simultaneously. The reduction of pressure decreased the required rod diameter to 2.12 in. Further reduction of the hydrostatic pressure to 0.433 psi to account for only 1 ft . head of head reduced the required shaft diameter to 0.71 in .

The stress and deflection of the plate for all three pressure conditions was also determined. Using the same calculation approach as before, a 0.04 in . rod was found to be sufficient for the $36,000 \mathrm{psi}$ yield strength. However, the calculated deflection was 17.22 in ., a value not possible for a 12 in. gate. Unlike the shaft, the required gate thickness would be driven by the resultant deflection. This was address by decreasing the allowable deflection to 0.01 in . (for minimal deflection), then calculating the required thickness to achieve it. The plate thickness required for the 0.01 in . deflection was approximately 0.50 in ., (see Appendix D, p 92 for calculations).

After observing the stress and strain behavior of the shaft and gate as isolated components, it was then necessary to observe them as a single system since the hand calculations had assumed the attaching component was fixed. Five configurations using standard shaft sizes and corresponding plate thicknesses based on key width were modeled in SOLIDWORKS for FEA analysis. Standard keyed shaft dimensions from 1 in. to 2-1/2 in. are shown below in Table 11.

Table 11. Standard Keyed Shaft Sizes (Grainger 2018)

| SHAFT DIAMETER (in.) |  | KEY WIDTH/PLATE <br> THICKNESS (in.) |  |
| :---: | :---: | :---: | :---: |
| Fraction | Decimal | Fraction | Decimal |
| 1 | 1 | $1 / 4$ | 0.25 |
| $11 / 4$ | 1.25 | $1 / 4$ | 0.25 |
| $11 / 2$ | 1.5 | $3 / 8$ | 0.375 |
| $13 / 4$ | 1.75 | $3 / 8$ | 0.375 |
| 2 | 2 | $1 / 2$ | 0.5 |
| 2 | $1 / 4$ | 2.25 | $1 / 2$ |
| 2 | $1 / 2$ | 2.5 | $5 / 8$ |

Gate Design Iteration 1: 1 in. Keyed Shaft with a 1/4 in. Thick Rectangular Plate
Model Details: A $12 \mathrm{in} . \mathrm{x} 10 \mathrm{ft}$ x $1 / 4 \mathrm{in}$. thick steel plate was fixed into the keyway of a $1 \mathrm{in} . \times 10-1 / 2 \mathrm{ft}$. A36 steel keyed shaft. The shaft was fixed from rotation at each end, and a uniform pressure of 0.433 psi was applied perpendicular to the plate.

FEA Results: Deflection at the center of the gate was approximately 1.76 in., (see Figure 51).

Remarks: The 1.76 in. deflection was considered too large.


Figure 51. Deflection of a 1 in. keyed shaft with a $1 / 4$ in. rectangular gate under 0.433 psi.

Gate Design Iteration 2: 1-1/4 in. Keyed Shaft with a 1/4 in. Thick Rectangular Plate
Model Details: A 12 in . x 10 ft x $1 / 4 \mathrm{in}$. thick steel plate was fixed into the keyway of a $1-1 / 4 \mathrm{in}$. x 10-1/2 ft. keyed A36 steel shaft. The shaft was fixed from rotation at each end, and a uniform pressure of 0.433 psi was applied perpendicular to the plate.

FEA Results: Deflection at the center of the gate was approximately 0.82 in., (see Figure 52).

Remarks: Deflection decreased from 1.76 in. to 0.82 in., but less deflection was desired.


Figure 52. Deflection of a 1-1/4 in. keyed shaft with a 1/4 in. rectangular gate under 0.433 psi

Gate Design Iteration 3: 1-1/2 in. Keyed Shaft with a 3/8 in. Thick Rectangular Plate
Model Details: A $12 \mathrm{in} . \times 10 \mathrm{ft}$ x $3 / 8 \mathrm{in}$. thick steel plate was fixed into the keyway of a $1-1 / 2 \mathrm{in}$. x $10-1 / 2 \mathrm{ft}$. A36 steel keyed shaft. The shaft was fixed from rotation at each end, and a uniform pressure of 0.433 psi was applied perpendicular to the plate.

FEA Results: Deflection at the center of the gate was 0.38 in., (see Figure 56).

Remarks: Deflection was decreased from 0.82 in. to 0.38 in., but team decided a deflection less than $1 / 4$ in. would be optimal.


Figure 53. Deflection of a 1-1/2 in. keyed shaft with a 3/8 in. rectangular gate under 0.433 psi

Gate Design Iteration 4: 1-3/4 in. Keyed Shaft with a 3/8 in. Rectangular Plate
Model Details: A 12 in . x 10 ft x $3 / 8 \mathrm{in}$. thick steel plate was fixed to the keyway of a $1-3 / 4 \mathrm{in}$. x $10-1 / 2 \mathrm{ft}$. A36 steel keyed shaft. The shaft was fixed from rotation at each end, and a uniform pressure of 0.433 psi was applied perpendicular to the plate.

FEA Results: Deflection at the center of the gate was 0.22 in., (see Figure 54).

Remarks: The results of the analysis showed the 1-3/4 in. shaft coupled with a $3 / 8$ in. thick plate was an optimal configuration. The 0.22 in . deflection of the gate was less than $1 / 4 \mathrm{in}$. and the $3 / 8$ in. gate plate thickness would remain consistent with the surrounding WISD plates. Also, the $9,600 \mathrm{psi}$ maximum von-Mises stress using the 1-3/4 in. keyed shaft and $3 / 8$ in. thick rectangular plate yields a factor of safety of 3.8, (see Figure 55).


Figure 54. Deflection of a 1-3/4 in. keyed shaft with a 3/8 in. rectangular gate under 0.433 psi


Figure 55. Von-Mises stress of a 1-3/4 in. keyed shaft with a 3/8 in. rectangular gate under 0.433 psi

After establishing a 1-3/4 in. keyed shaft and a $3 / 8$ in. thick gate plate would be adequate for the WISD prototype, one more design iteration was performed to determine if adding more support to each end of the shaft would further decrease the deflection by reducing the bending moment on the shaft. Custom shaft supports were designed to hinge each end of the shaft to the adjoining WISD plate. The addition of the two hinges did decrease the deflection approximately $10 \%$, from 0.22 in. to 0.20 in., but the maximum von-Mises stress increased from $9,600 \mathrm{psi}$ to $9,691 \mathrm{psi}$ (about $1 \%$ ). The sole advantage of acquiring custom machined supports would be the minimal 0.02 in . decrease in deflection. The team agreed pursuing the custom supports was not practical. However, an overview of the custom support design has been included below for future reference.

Gate Design Iteration 5: 1-3/4 in. Keyed Shaft, 3/8 in. Thick Rectangular Plate, Custom Hinges
Model Details: A 12 in . x 10 ft x $3 / 8 \mathrm{in}$. thick steel plate was fixed into the keyway of a $1-3 / 4 \mathrm{in}$. x $10-1 / 2 \mathrm{ft}$. A36 steel keyed shaft. Two custom machined hinges were fixed at each end of the gate plate to the adjoining plate of the WISD, (see Figure 56, Figure 57, and Figure 58). The shaft was fixed from rotation at each end, and a uniform pressure of 0.433 psi was applied perpendicular to the plate.

FEA Results: Deflection at the center of the gate was reduced to 0.20 in., (see Figure 59). Maximum von-Mises stress increased from to $9,691 \mathrm{psi}$, (see Figure 60)

Remarks: The cost associated with acquiring custom machined parts was not practical for the minimal 0.02 in. decrease in deflection. The concept was not pursued further.


Figure 56. Isometric view of custom hinges supporting closed gate.


Figure 57. Isometric view of custom hinges supporting open gate


Figure 58. Close-up of custom machined hinge (left) and installation (right).


Figure 59. Deflection of 1-3/4 in. shaft, 3/8 in. rectangular gate, and custom hinges under 0.433 psi.


Figure 60. Von-Mises stress of 1-3/4 in. shaft, $3 / 8$ in. thick gate, and custom hinges under 0.433 psi..

## Summary of Gate Design Iterations

The results of the four gate design iterations are shown below in Table 12.

Table 12. Summary of gate design iteration results

| ITERATION | SHAFT <br> DIAMETER | GATE <br> THICKNESS | MAX <br> STRESS | MAX <br> DEFLECTION |
| :---: | :---: | :---: | :---: | :---: |
| 1 | $1 \mathrm{in}$. | $1 / 4 \mathrm{in}$. | $54,890 \mathrm{psi}$ | 1.76 in. |
| 2 | $1-1 / 4 \mathrm{in}$. | $1 / 4 \mathrm{in}$. | $27,722 \mathrm{psi}$ | 0.818 in. |
| 3 | $1-1 / 2 \mathrm{in}$. | $3 / 8 \mathrm{in}$. | $15,762 \mathrm{psi}$ | 0.375 in. |
| 4 | $1-3 / 4 \mathrm{in}$. | $3 / 8 \mathrm{in}$. | $9,599 \mathrm{psi}$ | 0.217 in. |

## Overview of Structural Components Chosen for WISD Prototype

Based on the results of the previous sections, the structural basis of the WISD prototype will consist of the following A36 plain carbon steel components:

$$
\begin{array}{ll}
\text { QTY }=10 & 1-3 / 4 \mathrm{in} . \times 10-1 / 2 \mathrm{ft} . \text { long keyed shaft (gate system) } \\
\text { QTY }=10 & 10 \mathrm{ft} \times 10-7 / 16 \mathrm{in} . \times 3 / 8 \mathrm{in} . \text { plate (gate system) } \\
\text { QTY }=9 & 10 \mathrm{ft} \times 5 \mathrm{ft} \times 3 / 8 \mathrm{in} . \text { plate (plates after gate system) } \\
\text { QTY }=9 & 10 \mathrm{ft} \times 10 \mathrm{ft} \times 3 / 8 \mathrm{in} . \text { plate (plates before gate system) } \\
\text { QTY }=21 & 1 / 4 \mathrm{in.} \times 126 \text { in. threaded rod (plate support) } \\
\text { QTY }=21 & 1 / 4 \mathrm{in.} \mathrm{large} \mathrm{flange,} \mathrm{threaded} \mathrm{rivet} \mathrm{nut} \mathrm{(plate} \mathrm{support)}
\end{array}
$$



Figure 61. Simulation of chosen prototype components assembled within the WISD housing.

### 6.3 GATE MECHANISM

A key specification for the gate system was to ensure that all ten gates to open as quickly as possible. In similar applications, fast response requirements are typically met using clutches or brakes. The issue with using these devices for the WISD gate mechanism is they operate principally on friction and can momentarily inhibit movement until the device is completely disengaged. The rotating gate shafts require a more immediate release from their static position to develop the angular velocity necessary to satisfy the rapid response requirement.

A more suitable option was to use electromagnets. Electromagnets, primarily electromagnetic locks, are commercially available with various holding force capacities. These devices are low profile, easy to operate, and quick to respond. While connected to a power source, an electromagnet is held fixed to its respective strike plate with a predesignated amount of holding force. The fail-open mechanism allows the magnet to immediately disengage from the strike plate the moment power is removed from the magnet. Thus, a set of electromagnets affixed to each side of the gate shaft will hold all ten gates tightly closed while energized, then allow uninhibited, instantaneous rotation by simply interrupting the power source to the magnets.

To determine the total holding force required by each magnet, the 0.433 psi hydrostatic pressure of 1 ft . head of water was multiplied by the 12 in . height of the channel and 120 in . width of the gate. The result was a 624 lbf required holding force per channel, or 312 lbf holding force per magnet. However, to ensure a factor of safety of at least 2.0 , two standard electromagnet locks with 650 lbf holding force were specified for the final design, (see Appendix E, p 78).

The angular velocity of the shaft after the magnet disengages from the strike plate was also calculated using the 1 ft . head hydrostatic moment of $1248 \mathrm{lbf}-\mathrm{in}$ acting on the area of each plate and a shaft diameter of 1-3/4 in. The angular velocity was found to be 90.8 radians/second with a total travel time of 0.02 s for the entire 90-degree rotation, (see Appendix F, p 102 for calculation details). Initially, it was
thought a helical spring should also be incorporated into the gate mechanism to supplement the angular velocity of the shaft; however, hand calculations showed the moment created by 1 ft . of hydrostatic head will provide more than sufficient torque to quickly rotate the shaft.

The electromagnets were integrated with the final WISD design using the 3 in . extension already designed into each end of the gate shaft. A $3 / 8$ in. thick rectangular mounting plate was fixed to each 3 in. peripheral section assuming a welded connection, and the strike plates were attached to each mount using a single bolt. The electromagnet was secured to a secondary support system located on the outside of the WISD using machine screws. A 1 in. thick rubber damper was also installed on the secondary support system to help absorb the impact of the fast-moving gate as it reaches its final destination. Figure 62 and Figure 63 below provide a simulated overview of the electromagnet operation. The WISD housing, plates, and supports have been removed for clarity.


Figure 62. Simulation of the energized electromagnet holding gate in closed position.


Figure 63. Simulation of the de-energized electromagnet releasing gate to opening position.

It should also be noted that the recoil of the strike plate as it impacts the rubber stopper was assumed to be negligible since the 1 ft . deep column of water rushing over the open gate is not expected to allow the gate to lift once the shaft has rotated into its final position. Should this become an issue in the physical model, the rubber stopper on each end could be easily replaced with a second set of electromagnets.

## 7. DESIGN OF A $\mathbf{1 : 5}$ SCALE PHYSICAL MODEL

Testing and experimentation of the overall design of the full-size prototype began with the design of a $1: 5$ scale physical model of a single channel. All parameters of the WSID established in the preceding sections (including plate thickness, gate thickness, shaft size, and electromagnet holding force) were scaled proportionally to one-fifth of their original dimensions as follows, (see Table 13).

Table 13. WISD Structural Parameters for 1:5 Scale Model

| COMPONENT | ORIGINAL <br> PARAMETER | 1:5 SCALED <br> EQUIVALENT |
| :--- | :---: | :---: |
| Plate Width | 120 in. | 24 in. |
| Plate Length | 120 in. | 24 in. |
| Plate Thickness | 0.375 in. | 0.075 in. |
| Gate Width | 12 in. | 2.4 in. |
| Gate Length | 120 in. | 24 in. |
| Gate Thickness | $3 / 8 \mathrm{in}$. | 0.075 in. |
| Shaft Length | 126 in. | 25.2 in. |
| Shaft Diameter | $1-3 / 4 \mathrm{in}$. | 0.15 in. |
| Holding Force | 600 lbf. | 120 lbf. |

The concurrent team research on the time, velocity, and pressure required to accelerate the stagnate water in the reservoir section of the full-size prototype to the exit conduit would also be tested using the 1:5, single channel model. The prototype-model similarities for time, velocity, and pressure were found using the $1: 5$ model scale ratio with a Froude Model, (see Table 14).

Table 14. WISD Time, Velocity, and Pressure for 1:5 Scale Model

| PARAMETER | SIMILARITY MODEL <br> TO PROTOTYPE | 1:5 SCALED <br> EQUIVALENT |
| :--- | :---: | :---: |
| Time | $1 \mathrm{~s}=2.24 \mathrm{~s}$ | 0.38 s |
| Velocity | $1 \mathrm{ft} / \mathrm{s}=2.24 \mathrm{ft} / \mathrm{s}$ | $11.36 \mathrm{ft} / \mathrm{s}$ |
| Pressure | $1 \mathrm{psi}=5 \mathrm{psi}$ | 0.868 psi |

The body of the scaled physical model was made of clear plexiglass, allowing the team to observe the behavior of the water as it moved through the designed system. Plexiglass also enabled the installation of sensors on the outside of the model to record the velocity data of each test without visual obstruction.

### 7.1 THE 1:5 SCALE SECTIONAL MODEL

The 1:5 scale model was designed using a series of interchangeable sections to allow for flexibility in determining an optimal plate length for the target wave development. Each section includes square flanges attached to each end to bond the model together during testing. A basic layout of the sectional approach to the $1: 5$ scale model is shown in Figure 64.


Figure 64. Basic layout of $1: 5$ sectional model.

The three components shown in Figure 64 represent the inlet section, gate section, and outlet section. During testing, water would be added through an inlet located at the end of Component 1 until the capacity of Component 1 and Component 2 (up to the gate) is reached. Air pressure would then be applied to the body of water through another inlet, also located at the end of Component 1.

The experiment begins once power is cut to the electromagnets holding the gate in place. As the gate opens, water rushes forward to Component 3 under the force from the applied air pressure. The 45-degree angle inlet at Component 1 was one of three inlet angles shown to increase the vertical flow profile in the FLOW-3D simulation. Two more versions of Component 1 were designed to accommodate the two remaining angles of 25 -degrees and 35 -degrees for testing. Not shown in Figure 64 are the
additional sections designed to increase or decrease the lengths between Components for wave development. The sections range from 2.4 in . long to 24 in . long to simulate between 1 ft . and 10 ft . in the prototype, respectively. Complete drawings for this system can be found in Appendix H, p 113.

### 7.2 THE PIPE MODEL

FLOW-3D simulations of the 1:5 scaled model showed that the scale model could produce a nearvertical flow profile similar to the prototype if adequate pressure between 3 and 5 psi was quickly applied to the stagnant body of water. However, the mechanics of rapid air application had not yet been established by the team. An inlet port had been added to the sectional model for the air to enter, but the expansion of compressed air, the pressure loss across the pipe leading from air compressor to inlet, and other minor losses needed to be resolved prior to constructing the sectional model for testing.

A simple device for measuring pressure and velocity was devised. A pipe model would be constructed in the form of a u-tube using 4 in. NPS Schedule 40 clear PVC pipe, (see Figure 65). The u-tube shape allowed water to rest in the bottom of the device without the need for gates using a 45 -degree angled inlet and outlet. The water would be filled to a depth of 1 ft . to simulate the head of a prototype channel, and the pressure to accelerate this water to a steady $11.36 \mathrm{ft} / \mathrm{s}$ velocity using compressed air could be investigated.


Figure 65. Pipe model for pneumatic testing.

## 8. PNEUMATICS

The decision to use a pneumatic system to accelerate the column of water in the reservoir section of the WISD to the required velocity of $25.4 \mathrm{ft} / \mathrm{s}$ before exiting the conduit was based on the research and recommendations of 2016 WISD member, Gregory Roberts. For a $1: 5$ scale model of the full-size prototype, the required velocity would be decreased to $11.36 \mathrm{ft} / \mathrm{s}$, as previously shown in Table 14. The pressure given by FLOW-3D to achieve a constant $11.36 \mathrm{ft} / \mathrm{s}$ velocity be the exit of the scale model was 3 psi to 5 psi , depending on the inlet angle.

The initial length of the pipe model was designed to be 10 ft . overall, approximately 5 ft . for the u tube section and 5 ft . for the horizontal outlet. The team agreed that this length could be adjusted as required. The air compressor specified for testing was a 60 -gallon tank rated for 9.0 scfm at 100 psi with a standard $1 / 4 \mathrm{in}$. outlet. The pipe model was designed for a 4 in . NPS Schedule 40 clear PVC pipe with an inner diameter of 4.026 in . To minimize abrupt changes in pipe diameter from the $1 / 4 \mathrm{in}$. outlet of the compressor to the 4 in. PVC, the team chose a 2 in . NPS Schedule 40 PVC pipe to couple the two lines.

The FLOW-3D model also showed that the air pressure would need to be applied, then removed in approximately 0.38 seconds for a steady $11.36 \mathrm{ft} / \mathrm{s}$ velocity wave to develop. The valves controlling the pressure would therefore need to be fast-acting, actuated valves as manually opening and closing the valves could not be accomplished in the necessary timeframe. Solenoid valves with response times less than 50 milliseconds were chosen for the task. The following diagram illustrates the concept chosen by the team for applying the air pressure to the pipe model, (see Figure 66). Valves 1 and 3 are normally open (NO) solenoid valves that close when energized. Valve 2 is a normally closed (NC) solenoid valve that opens when energized. Power to all three valves is controlled by a multifunction timer relay.


Figure 66. Diagram of pneumatic system for pipe model.

At $\mathrm{t}<0$ seconds, all three valves are in their normal state (unenergized). The compressor is turned on to allow steady state air flow to develop through the unenergized NO Valve 1. The section of pipe from Inlet/Outlet 2 to the pipe model is open to atmosphere since NO Valve 3 is also unenergized. At $t=0$ seconds, all three valves become energized using the timer relay. NO Valve 1 closes, NC Valve 2 opens, and NO Valve 3 closes. Flow is directed from the air compressor through Valve 2 to the pipe model. At $\mathrm{t}=0.38$ seconds, the timer relay opens the circuit to all three valves and they return to their normal unenergized state. Air flow is directed back to the open Valve 1 since Valve 2 is now closed. Valve 3 allows atmospheric pressure to enter the pipe model so that a vacuum does not develop behind the moving column of water as the air pressure is removed. A simulated model using SOLIDWORKS is shown for the entire system below, (see Figure 67).


Figure 67. SOLIDWORKS simulation of complete pipe model assembly.

The team was able to locate three $2 \mathrm{in} .120 \mathrm{~V} / 60 \mathrm{~Hz}$ rated for $5-100$ psig that would meet the requirement of a less than 50 millisecond response time; however, the valves were not cost effective for the WISD budget. The 5 psig minimum inlet pressure was also a concern as it limited the team's testing range. Valve 1 would always receive the required minimum 5 psig when in use, but the other two valves would receive less than required 5 psig when Valve 1 was open. If Valve 3 does not re-open after $t=0.38$ seconds, a vacuum could develop behind the water column in the tube, slowing or even stopping water flow.

The pipe size connecting the compressor to the model was then reduced to a 1 in . NPS Schedule 40 PVC to enable 1 in . valve connections. The 1 in . valves had the same reaction time as the 2 in . valves but were less expensive and rated for $0-125 \mathrm{psi}$. Then after reviewing the design, it was noted that the only valve requiring a larger orifice for flow control was Valve 2. Valve 1 and 3 were only in place to direct the flow to Valve 2 or atmosphere. Therefore, Valves 1 and 3 were further reduced to $1 / 2 \mathrm{in} .120 \mathrm{~V} / 50 \mathrm{~Hz}$ solenoid valves with a $0-150$ psi pressure rating. The team agreed to leave the pipe size unchanged from 1 in . and use reducers to connect the $1 / 2 \mathrm{in}$. valves.

## 9. RECOMMENDATIONS FOR FUTURE WORK

The construction of the pipe model is currently underway at CFEL. Experimentation with the pipe model will allow the WISD team to gather data for various configurations of input pressure and solenoid opening time, then determine the optimal combination required to produce the velocity, wave profile, and wave development length anticipated in the $1: 5$ scale model. If the team can recreate the same output conditions in the $1: 5$ scale model using the relationship established in the pipe model, the pneumatic approach discussed in this work should be similarly applicable for a full-size prototype design.

One of the largest obstacles facing the team throughout this investigation was the difficulty in creating a near-vertical flow profile. In each FLOW 3-D simulation, the flow was observed to move more rapidly in the bottom portion of the wave than it did near the top. The team understood this to be a logical behavior due to gravity that would eventually need to be revisited. As such, a second recommendation for future work would be the refinement of the outlet conduit. Several roughening materials, such as sandpaper, were discussed as potential candidates; however, the team was unable to study these effects further due to the time constraints of the project. Experimentation with various materials may provide the friction necessary for slowing the progression of the lower portion of the wave, thereby achieving a more accurate vertical wave profile.

Finally, the results discussed throughout the course of this thesis are based solely on computational modeling. The physical 1:5 scale model and full-size prototype will undoubtedly require modifications. In general, the WISD team recommends approaching the construction of a physical model in a manner similar to the methodology used here. First, establish the strength of materials to safely achieve a working system. Second, ensure that modifications to any component will not adversely impact the function of another. And third, experiment on a smaller scale prior to making any adjustments to the larger physical system.

## 10. CONCLUSION

A prototype for the Wave Impact Simulation Device (WISD) has been designed to meet the requirements of a near vertical wave profile, a uniform velocity of $25.4 \mathrm{ft} / \mathrm{s}$, and a rapid-response release system. The prototype will incorporate a pneumatically driven, ten level water system. The ten channels of the prototype will be constructed of 10 ft . x 10 ft x $3 / 8 \mathrm{in}$. thick, A36 steel plates supported by interspersed $1 / 4 \mathrm{in}$. steel rod and rivet nut supports. The maximum deflection under a uniform 1.3 psi pressure (equivalent to 3 ft . of hydrostatic head) was found to be 0.04 in . using the SOLIDWORKS Simulation FEA. The gates of the prototype consist of $1-3 / 4 \mathrm{in}$. keyed shafts with $3 / 8 \mathrm{in}$. thick steel, rectangular plates spanning the width of each channel. Electromagnets were chosen as the preferred gate mechanism to achieve a $90.8 \mathrm{rad} / \mathrm{s}$ angular velocity and 90 -degree travel time of 0.02 seconds.

All design requirements and resultant parameters from the WISD prototype simulation study were reduced to $1: 5$ of their respective value to accommodate scale-model testing. The scale model will allow experimentation on the proposed prototype design to verify and/or modify parameters as necessary, prior to the construction of a full-size system. A complete set of drawings for one level of the $1: 5$ scale model was included in this work. To establish the pneumatic requirements for achieving a near vertical, constant velocity wave profile, a third model was designed to study the effects of compressed air and response time of fast-acting solenoid valves. This model, referred to as the pipe model, is currently under construction at CFEL. Once the relationship between input pressure, valve opening time, velocity, wave profile, and wave development length is established, the same pneumatic approach will be applied to the $1: 5$ scale model, and eventually, the full-size WISD prototype.

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APPENDIX A: Hand Calculations for Required Plate Thickness
CFEL-W 5/15/17 Hand Calculations Larinda Nichols 1/9 Wave Machine


Current Assumptions
Overall Dimensions:
Plate length behind gates $=10 \mathrm{ft}$
End angle for each plate $=45^{\circ}$
Plate length after gates $=5 \mathrm{ft}$
Overall exit area $=10 \mathrm{ft} \times 10 \mathrm{ft}$
Number of plates $=9$
Surrounding concrete $=1 \mathrm{ft}$ thick
Volume of entire water column:
Without plates $=10 \mathrm{ft} \times 10 \mathrm{ft} \times 10 \mathrm{ft} \approx 1000 \mathrm{ft}^{3}$
Pressure at bottom of column behind gates:

$$
\begin{aligned}
& P=\gamma h \\
& \gamma_{\mathrm{H}_{2} \mathrm{O}}\left(60^{\circ} \mathrm{F}\right)=62.4 \mathrm{lbf} / \mathrm{ft}^{3} \\
& h_{\mathrm{H}_{2} \mathrm{O}}=10 \mathrm{ft} \\
& P_{\text {Bottom }}=624 \frac{\mathrm{lbf}}{\mathrm{ft}^{2}}\left(\frac{1 \mathrm{ft}^{2}}{144 \mathrm{in}^{2}}\right)=4.33 \mathrm{psi}
\end{aligned}
$$

## From Greg Robert's Simulation:

- 10 Gate Components
- Horizontal Rotation ( $90^{\circ}$ about $y$-axis)
$-\omega= \pm 15.708 \mathrm{ft} / \mathrm{s}$
-0.1 sec .
Target wave speed $=25.4 \mathrm{ft} / \mathrm{s}$
Current design Maximum Velocity $=24.8259$ to 25.4243 fls
* Simulation time of maximum Velocity $\approx 0.621$ to 0.642 s
[see Table 2 for more information (Roberts, Thesis)


## Project Considerations

1. Gate mechanism
2. Leakage
3. Plate Weight*
4. Maintenance access
5. Air pressure to displace fluid
6. Ease of plate insertion/removal
7. Fluid reservoir
8. Power required to operate

9 Flow channel
10. Material \& Cost
II. Observation area
$\left[\begin{array}{c}\text { * Material assumptions } \\ \text { - Concrete housing } \\ \text { - Steel plates } \\ \text { Steed gates }\end{array}\right]$

Gate Concepts

1. Bottom hinged

2. Butterfly valve


Gate Considerations

- Rotational inertia
- Time to rotate open
- Force to rotate open
- Force to usist water column
- Mechanism for action
- Stress/strain/fatigue on gate and mechanism
- Water profile disruption

Forces on Gates (Hydrostatic Magnitude)


- Gate is 10 ft . Wide and $\sim 1 \mathrm{ft}$. Tall
- $\gamma=62.4 \mathrm{lb} / \mathrm{ft}^{3}$
- Possible Gate of 5 ft . Wide and $\sim 1 \mathrm{ft}$ Tall
- Linear Distribution

Individual Gates: $F_{H}$ acts at $2 / 3 H \quad\left(y_{R}=2 / 3 H\right)$


$$
F_{1 H}=P_{\text {ave }} \cdot A
$$

$$
H_{c}=1 / 2 \mathrm{H} \text { (Rectangular Gate) }
$$

$$
P_{\text {ave }}=\gamma_{\mathrm{H}_{2} \mathrm{O}} H_{c}
$$

Bottom Hinged Gate


Torque Required at hinge for $\sum M_{H}=0$ is $1170.78 \mathrm{lb}-\mathrm{ft}$ Center Hinged Gate


Again, assume bottom gate, 5 ft wide
$A=5 \mathrm{ft}^{2}, \quad y_{c}=9.5 \mathrm{ft}$
$F_{R}=2964 \mathrm{lb} \quad$ (same as above)
$y_{R}=9.605 \mathrm{ft}$ (same as above)
But now the tor que required for $\sum M_{H}=0$,

$$
\begin{aligned}
& T_{H}=F_{R} L_{R}=(29641 \mathrm{lb})\left(y_{R}-y_{C}\right) \\
& T_{H}=(2964 \mathrm{lb})(9.605 \mathrm{ft}-9.5 \mathrm{ft}) \\
& T_{H}=(2964 \mathrm{lb})(0.105) \\
& T_{H}=311.22 \mathrm{lb}-\mathrm{ft}
\end{aligned}
$$

## Plate Analysis

Assumption: Plates will be 5 ft wide
For plates 15 feet in length


Carbon Steel (shia)

$$
\begin{aligned}
& W=0.282 \mathrm{lbg} / \mathrm{m}^{3} \\
& E=30.0 \mathrm{Mpsi} \\
& G=11.5 \mathrm{Mpsi}
\end{aligned}
$$

$$
v=0.292
$$

(Ib)

Weight/sheet
$0.282 \mathrm{lg} / \mathrm{in}^{3}(180 \mathrm{in})(60 \mathrm{in})(0.1875 \mathrm{in})=571$
$0.282 \mathrm{~kg} / \mathrm{in}^{3}(180 \mathrm{in})(60 \mathrm{in})(0.2 \mathrm{sin}) \equiv 762$
$0.282 \mathrm{~kb} / \mathrm{in}^{3}(180 \mathrm{in})(60 \mathrm{in})(0.37 \mathrm{sin})=1142$
$0.282 \mathrm{lb} / \mathrm{in}^{3}(180 \mathrm{in})(60 \mathrm{in})(0.5 \mathrm{in})=1523$

* Least amt of weight $\approx 571 \mathrm{lb} /$ sheet (for $15^{\prime} \times 5^{\prime} \times 3 / 6^{\prime \prime}$ )
* Total number of sheets needed $=2 /$ level $\times 9$ level o $=18$

For plates $10^{\prime} \times 5^{\prime}$

$\xrightarrow{\text { Area of each plate }} A=120^{\prime \prime} \times 60^{\prime \prime}=7200 \mathrm{in}^{2}$
Total number of plates $3 /$ level $\times 9$ levels $=27$

| Weight per Plate |
| :--- |
| $1 / 16$ in. |
| $\approx 381 \mathrm{l}$. |
| 50816 | $1 / 4 \mathrm{in} . \approx 508 \mathrm{lb}$

$3 / 8 \mathrm{in} \approx 762 \mathrm{lb}$. $3 / 8 \mathrm{in} . \approx 762 \mathrm{lb}$.
$1 / 2 \mathrm{in}$.

* Need to analyze: Force, Deflection, Stress, and slope to determine thickness require dor application.
Assume simply supported:

L. Nichols

Forces, Deflection and Stress on Plates
Analysing bottom plate (plate \#9) since the force will lee greatest here.

$$
\begin{aligned}
& P=\gamma h=\left(62.4 \mathrm{lb} / \mathrm{ft}^{3}\right)(9 \mathrm{ft})=561.6 \mathrm{1b} / \mathrm{st}^{2}\left(\frac{1 \mathrm{ft}^{2}}{144 \mathrm{~m}^{2}}\right)=3.9 \mathrm{psi} \\
& A=\left(10^{\prime} \times 5^{\prime}\right)=(120 \mathrm{in} \times 60 \mathrm{in})=7200 \mathrm{in}^{2}
\end{aligned}
$$

Calculations Using "Rank's Formulas for Stress \& Strain" - CHII, TABLE II .4, Formulas for Flat Plates with
straight Boundaries 8 Cont. Thickness
Case No. 2. Rectangular Plate; three edges simply supported, one edge free


2a. Loading uniform over entice plate
2b. Loading uniformly increasingalong a
(These calculations assume $\nu=0.3$, steel)
For plates $A$ and $B\left(10^{\prime} \times 5^{\prime}\right)$

$$
\begin{aligned}
& a=10 \mathrm{ft}=120 \mathrm{in} \\
& b=5 \mathrm{ft}=60 \mathrm{in} \\
& \text { Case } 2 a . \text { (Uniform) }
\end{aligned}
$$

For plate $C\left(5^{\prime} \times 10^{\prime}\right) a=5 \mathrm{ft}=60$ in

$$
b=10 \mathrm{ft}=120 \mathrm{in}
$$

Case Rb. (Increasing)
2a. $\sigma_{\max }=\frac{\beta q b^{2}}{t^{2}} \quad y_{\max }=\frac{-\alpha q b^{4}}{E^{3}} \quad$ from table,

$$
\left.\frac{a}{b}=2\right\} \begin{aligned}
& \beta=0.79 \\
& \alpha=0.165
\end{aligned}
$$

$$
q=3.9 \text { psi (plate } * 9)
$$

For $3 / 16^{\prime \prime}$ thick: $\quad \sigma_{\text {max }}=\frac{(0.79)\left(3.9 \mathrm{lg} / \mathrm{in}^{2}\right)(60 \mathrm{in})^{2}}{(0.1875)^{2}}=315,495$ psi

$$
n=\frac{S_{y t}}{\sigma_{\max }}=\frac{57,990 p \mathrm{pi}}{315,495 p s i}=0.565
$$

Factor of safety is $\angle 1$, no need to proceed further with 3016 " thick plate' (unkss we use something other than plain carton steed.
(Rock's Formulas Continued)
For $3 / 8^{\prime \prime}$ thick: $\sigma_{\text {max }}=\frac{(0.79)\left(3.9 \mathrm{hb} / \mathrm{rin}^{2}\right)(60 \mathrm{in})^{2}}{(0.375 \mathrm{in})^{2}}=78,873.6 \mathrm{psi}$

$$
n=\frac{S_{y t}}{\sigma_{\text {max }}}=\frac{57,990 p \text { si }}{78,873.6 p s e}=0.735:(\text { less }<1)
$$

For $1 / 2^{\prime \prime}$ thick: $\sigma_{\text {max }}=\frac{(0.79)\left(3.916 / \mathrm{m}^{2}\right)(60 \mathrm{in})^{2}}{(0.5 \mathrm{in})^{2}}=44366.4 \mathrm{psi}$

$$
n=\frac{S y_{t}}{\sigma_{\max }}=\frac{57,990 \text { psi }}{44,366.4 \text { psi }}=1.31 \quad \begin{aligned}
& \text { (factor of } \\
& \text { safety) }
\end{aligned}
$$

Deflection using $1 / 2^{\prime \prime}$ thick carton steel:

$$
y_{\text {max }}=\frac{-\alpha q b^{4}}{E z^{3}}=\frac{(-0.165)\left(3.966 / \mathrm{in}^{2}\right)(60 \mathrm{in})^{4}}{\left(30.46 E 664 / \mathrm{in}^{2}\right)(0.5 \mathrm{in})^{3}}=-2.2 \mathrm{in} .
$$

Max deflection $\approx-2.2$ inches
For plate $C, 1 / 2^{\prime \prime}$ thick carbon steel (plain)

$$
\text { 2b. } \begin{aligned}
\sigma_{\text {max }} & =\frac{\beta q b^{2}}{t^{2}}, y_{\text {max }}=-\frac{\alpha q b^{4}}{E t^{3}} \quad \begin{array}{l}
\text { from table, } \\
\left.\frac{a}{b}=0.5\right\} \quad \beta=0.11 \\
\alpha=0.026
\end{array} \\
\sigma_{\text {max }} & =\frac{(0.11)\left(3.9 \mathrm{~b} / \mathrm{m}^{2}\right)(120 \mathrm{in})^{2}}{\left(0.5 \mathrm{in}^{2}\right.}=24.710 .4 \text { poi } \quad(n=2.34) \\
y_{\text {max }} & =\frac{(-0.026)\left(3.9 \mathrm{~b} / \mathrm{in}^{2}\right)(120 \mathrm{in})^{4}}{\left(30.46 E 6 \mathrm{bb} \mathrm{in}^{2}\right)(0.5 \mathrm{in})^{3}}=-5.52 \mathrm{in}!
\end{aligned}
$$

- Deflection is almost $1 / 2$ the height of the 1 ft opening.
- Need to look into stronger/stiffor steel, or maybe "fix." plates to the sides of the channel.
Option 1: "Fix" plates on sides of flume, other 2 sides
"simply" supported
Plates $A+B \rightarrow$ Case 4


Plate $C \rightarrow$ Case 5


Option 2: Stronger/Stiffer Steel

Cold Rolled?
Galvanized?
Option 3: Decrease width/span of plates

$$
\text { Maybe } 2.5^{\prime} \times 10^{\prime} \text { plates? }
$$

First, looking at option 1: Additional Support
Fixed along sides of channel:


Case 4: $\left.\sigma_{\text {max }}=\frac{\beta q b^{2}}{t^{2}}, \quad y_{\text {max }}=-\frac{\alpha q b^{4}}{E t^{3}} \quad \frac{a}{b}=2\right\}_{\alpha}^{\beta=0.73} \begin{aligned} & \alpha=0.054\end{aligned}$
For $3 / 16^{" 1}$ thick: $\sigma_{\text {max }}=\frac{(0.73)\left(3.916 / /^{2}\right)(60 \mathrm{im})^{2}}{(0.1875 \mathrm{im})^{2}}=291,532.8$

- Doesn't decrease $\sigma_{\text {max }}$ by much. (originally 315,495 psi) Next, try decreasing width/span of plates (option 3)

For plates before gates open:


Case Ra: $\left.\sigma_{\text {max }}=\frac{\beta q b^{2}}{E^{2}}, \quad y_{\text {max }}=\frac{-\alpha a b^{4}}{E t^{3}} \quad \frac{a}{b}=4\right\}^{\beta=0.80} \begin{aligned} & \alpha=0.167\end{aligned}$
$\left[3 / 16^{\prime \prime}\right.$ thick $] \quad \sigma_{\text {max }}=\frac{(0.8)\left(3.9 \mathrm{ld} / \mathrm{n}^{2}\right)\left(3 \mathrm{D}_{\mathrm{in}}\right)^{2}}{(0.1875 \mathrm{in})^{2}}=79,872 \mathrm{psi}$

$$
n=\frac{S_{y t}}{\sigma_{\text {max }}}=\frac{57,990 p s i}{79,872 p s i}=0.726 Y(\text { less }<1)
$$

$$
\begin{aligned}
& \text { For } 3 / 8^{\prime \prime} \rightarrow \sigma_{\text {max }}= \frac{(0.8)\left(3.916 / \mathrm{m}^{2}\right)(30 \mathrm{in})^{2}}{(0.375 \mathrm{in})^{2}}=19,968 \mathrm{psi} \\
& n=\frac{S y t}{\sigma_{\text {max }}}=\frac{57,990 p \mathrm{pi}}{79,968 p \mathrm{pi}}=2.9 \mathrm{p} \\
& y_{\text {max }}=-\frac{\alpha q b^{4}}{E^{3} t^{3}}=\frac{(-0.167)(39, \mathrm{~b})(30 \mathrm{n})^{4}}{\left(3046 E 664 \mathrm{in}^{2}\right)(0.375 \mathrm{in})^{3}}=-0.33 \mathrm{in}
\end{aligned}
$$

This configuration would work best if possible.
Factor of Safety $=2.9$
Max Deflection $=0.33$ in
Weight of $2.5^{\prime} \times 10^{\prime} \times 3 / 8^{\prime \prime}$ plates $=380.7 \mathrm{lb}$

* Issue for plates after gate section?

To discuss: Plate options before gate section Plate option after gate section Options for steel Additional support for plates

CFEL-W 6/28/17 Hand Calculations Lavinda Nichols 1/1 New Geometry: $4-30 \mathrm{in}$. Plates per Level ( 10 ft .Plates) Plate thickness $=3 / 16 \mathrm{in}$.

$$
\begin{gathered}
\text { Individual Plate Weight }=\left(0.282 \mathrm{~kb} / \mathrm{in}^{3}\right)(120 \mathrm{in})(30 \mathrm{in})(0.1875 \mathrm{in}) \\
\text { Weach }=190.35 \mathrm{lb}
\end{gathered}
$$

Roark's Maximum Formulas (6)

$$
\begin{gathered}
\sigma_{\text {max }}=\frac{-(0.5)(3.9 p s i)(30 \mathrm{in})^{2}}{(0.1875 \mathrm{in})^{2}}=49.920 \text { psi } \\
n=\frac{S_{y t}}{\sigma_{\text {max }}}=\frac{57,990 p \mathrm{pi}}{49,920 p \mathrm{si}}=1.16 \\
y_{\text {max }}=\frac{-(0.0285)\left(3.9^{\left.\mathrm{lb} / \mathrm{in}^{2}\right)(30 \mathrm{in})^{4}}\right.}{\left(30.46 \mathrm{E} 6 \mathrm{lb} \mathrm{in}^{2}\right)(0.1875) \mathrm{in}^{3}}=0.44839 \mathrm{in}
\end{gathered}
$$

WISD - ROARKS FORMULAS FOR PLATE STRESS AND DEFLECTION

|  |  |  |  | Width (ft) | Width (in) | Length (ft) | Length (in) |  |  | (in) | (in) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Channel Height ( ft ) | 1 |  |  | $b$ | b | a | a |  | Thickness 1 | $3 / 8$ | 0.3750 |
| Channel width ( ft ) | 10 |  | Geometry 1 | 10 | 120 | 10 | 120 |  | Thickness 2 | 1/4 | 0.2500 |
| Specific Density Water ( $\mathrm{lb} / \mathrm{ft}^{2}$ ) | 62.4 |  | Geometry 2 | 5 | 60 | 10 | 120 |  | Thickness 3 | $3 / 16$ | 0.1875 |
| Pressure on Plate (psi) | 0.433 |  | Geometry 3 | 2.5 | 30 | 10 | 120 |  | Thickness 4 | $1 / 8$ | 0.1250 |
| Roark's tabulated values | a/b | 1 | 1.2 | 1.4 | 1.6 | 1.8 | 2 | $\infty$ |  |  |  |
|  | $\beta$ | 0.4182 | 0.4626 | 0.486 | 0.4968 | 0.4971 | 0.4973 | 0.5000 |  |  |  |
|  | $\alpha$ | 0.021 | 0.0243 | 0.0262 | 0.0273 | 0.028 | 0.0283 | 0.0285 |  |  |  |


| Elastic Modulus (psi) | $30,457,924.91$ |
| :--- | :--- |
| Poisson's Ratio | 0.3 |
| Yield strength (psi) | $31,994.45$ |
| Mass Density $\left(\mathrm{lb} / \mathrm{in}^{3}\right)$ | 0.281793 |
| CASE No. $6-$ MAX YIELD | CASE NO. $6-$ MAX DEFLECTION |
| Max $\sigma=\frac{-\beta q b^{2}}{t^{2}}$ | Max $y=\frac{-a q b^{4}}{E t^{3}}$ |


| $10^{\prime} \times 10^{\prime}$ GEOMETRY | width (in) | Length <br> (in) | $\begin{aligned} & \text { Ratio } \\ & (\mathrm{w} / 1) \end{aligned}$ | Tabulsted Value | Tabulated value | Pressure (psi) | Thickness (inches) | volume $\left(\right.$ in $\left.^{2}\right)$ | weight (IB) | Max Yield (psi) | Factor of Safety | Max Deflect <br> [in) | Max Defiect vs. |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | b | a | a/b | $\beta$ | $\alpha$ | 9 | t | v | w | $\sigma$ | N | y | Thickness |
| $3 / 8{ }^{\prime \prime}$ | 120 | 120 | 1 | 0.4182 | 0.021 | 0.433 | 0.3750 | 5400 | 1521.68 | 18556.93 | 1.7 | 1.1748 | 3.1 |
| $1 / 4^{\prime \prime}$ | 120 | 120 | 1 | 0.4182 | 0.021 | 0.433 | 0.2500 | 3600 | 1014.45 | 41753.09 | 0.8 | 3.9650 | 16 |
| 3/16" | 120 | 120 | 1 | 0.4182 | 0.021 | 0.433 | 0.1875 | 2700 | 760.84 | 74227.71 | 0.4 | 9.3986 | 50 |
| $1 / 8{ }^{\prime \prime}$ | 120 | 120 | 1 | 0.4182 | 0.021 | 0.433 | 0.1250 | 1800 | 507.23 | 167012.4 | 0.2 | 31.7202 | 254 |
| $5^{\prime} \times 10^{\prime}$ GEOMETRY | width <br> (in) <br> b | Length <br> (in) <br> a | Ratio <br> (w/l) <br> a/b | Tabulsted Value $\beta$ | Tabulated Value $\alpha$ | Pressure (psi) q | Thickness (inches) t | volume [in ${ }^{2}$ ) v | weight <br> (IB) <br> w | Max Yield (psi) <br> $\sigma$ | Factor of Sofety N | Max Deffect <br> (in) <br> y | Max Defiect vs. Thickness |
| $3 / 8{ }^{\prime \prime}$ | 60 | 120 | 2 | 0.4973 | 0.0283 | 0.433 | 0.3750 | 2700 | 760.84 | 5516.715 | 5.8 | 0.0990 | 0.3 |
| $1 / 4{ }^{\text {" }}$ | 60 | 120 | 2 | 0.4973 | 0.0283 | 0.433 | 0.2500 | 1800 | 507.23 | 12412.61 | 2.6 | 0.3340 | 1.3 |
| $3 / 16^{\prime \prime}$ | 60 | 120 | 2 | 0.4973 | 0.0283 | 0.433 | 0.1875 | 1350 | 380.42 | 22066.86 | 1.4 | 0.7916 | 4.2 |
| $1 / 8^{\prime \prime}$ | 60 | 120 | 2 | 0.4973 | 0.0283 | 0.433 | 0.1250 | 900 | 253.61 | 49650.43 | 0.6 | 2.6717 | 21 |
| $2.5{ }^{\prime} \times 10^{\prime}$ GEOMETRY | width <br> (in) <br> b | length <br> (in) <br> a | Retio <br> (w/1) <br> $a / b$ | Tabulated Value $\beta$ | Tebulated Value $\alpha$ | Pressure (psi) q | Thickness (inches) $t$ | volume $\left(\right.$ in $\left.^{2}\right)$ v | weight <br> (IB) <br> w | Max Yield (psi) | Fsctor of Sstety N | Max Deflect <br> (in) <br> Y | Max Defiect <br> vs. <br> Thickness |
| $3 / 8^{\prime \prime}$ | 30 | 120 | 4 | 0.5000 | 0.0285 | 0.433 | 0.3750 | 1350 | 380.42 | 1386.667 | 23.1 | 0.0062 | 0.02 |
| $1 / 4^{\prime \prime}$ | 30 | 120 | 4 | 0.5000 | 0.0285 | 0.433 | 0.2500 | 900 | 253.61 | 3120 | 10.3 | 0.0210 | 0.08 |
| 3/16 ${ }^{\text {n }}$ | 30 | 120 | 4 | 0.5000 | 0.0285 | 0.433 | 0.1875 | 675 | 190.21 | 5546.667 | 5.8 | 0.0498 | 0.27 |
| $1 / 8^{\prime \prime}$ | 30 | 120 | 4 | 0.5000 | 0.0285 | 0.433 | 0.1250 | 450 | 126.81 | 12480 | 2.6 | 0.1682 | 1.35 |
| GEOMETRIES MEETING ROARK'S RESTRICTIONS | Width <br> (in) <br> b | Length <br> (in) <br> a | Ratio <br> (w/l) <br> a/b | Tobulated Value B | Tabulated Value $\alpha$ | Pressure (psi) 9 | Thickness (inches) $t$ | volume $\left(\mathrm{in}^{2}\right)$ v | weight (ib) w | Max Yield (psi) a | Fsctor of 5stety N | Max Deflect <br> (in) <br> $y$ | Max Deflect <br> vs. <br> Thickness |
| $3 / 8^{\prime \prime}-5^{\prime} \times 10^{\prime}$ Plate | 60 | 120 | 2 | 0.4973 | 0.0283 | 0.433 | 0.3750 | 2700 | 760.84 | 5516.715 | 5.8 | 0.0990 | 0.26 |
| $3 / 8^{\prime \prime}-2.5^{\prime} \times 10^{\prime}$ Plate | 30 | 120 | 4 | 0.5000 | 0.0285 | 0.433 | 0.3750 | 1350 | 380.42 | 1386.667 | 23.1 | 0.0062 | 0.02 |
| $1 / 4^{\prime \prime}-2.5^{\prime} \times 10^{\prime}$ Plate | 30 | 120 | 4 | 0.5000 | 0.0285 | 0.433 | 0.2500 | 900 | 253.61 | 3120 | 10.3 | 0.0210 | 0.08 |
| $3 / 16^{\prime \prime}-2.51 \times 10^{\prime}$ Plate | 30 | 120 | 4 | 0.5000 | 0.0285 | 0.433 | 0.1875 | 675 | 190.21 | 5546.667 | 5.8 | 0.0498 | 0.27 |

COMPARISON OF ROARK RESULTS WITH SOLIDWORKS

|  | MAXIMUM YIELD STRESS (psi) |  |  | MAXIMUM DEFLECTION (in) |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | ROARK's | souidworks | \% diff | ROARK's | SOLDWORKS | \% diff |
| $3 / 8^{\prime \prime}-5^{\prime} \times 10^{\prime}$ Plate | 5516.7 | 5108.5 | 7.7 | 0.0990 | 0.1110 | 11.5 |
| $3 / 8^{\prime \prime}-2.5^{\prime} \times 10^{\prime}$ Plate | 1386.7 | 1190.5 | 15.2 | 0.0062 | 0.0069 | 10.9 |
| $1 / 4^{\prime \prime}-2.55^{\prime} \times 10^{\prime}$ Plate | 3120.0 | 2688.7 | 14.9 | 0.0210 | 0.0234 | 10.5 |
| $3 / 16^{\prime \prime}-2.51 \times 10^{\prime}$ Plate | 5546.7 | 4703.0 | 16.5 | 0.0498 | 0.0547 | 9.3 |

## APPENDIX B: SOLIDWORKS Simulation Settings

Study Properties

| Analysis type | Static |
| :--- | :--- |
| Mesh type | Solid Mesh |
| Thermal Effect: | On |
| Thermal option | Include temperature loads |
| Zero strain temperature | 298 Kelvin |
| Include fluid pressure effects from | Off |
| Solver type | FFEPlus |
| Inplane Effect: | Off |
| Soft Spring: | Off |
| Inertial Relief: | Off |
| Incompatible bonding options | Automatic |
| Large displacement | Off |
| Compute free body forces | On |
| Friction | Off |
| Use Adaptive Method: | Off |

Mesh information

| Mesh type | Solid Mesh |
| :--- | :--- |
| Mesher Used: | Curvature-based mesh |
| Jacobian points | 29 Points |
| Maximum element size | 1.20561 in |
| Minimum element size | 0.0602806 in |
| Mesh Quality | High |

Mesh information - Details

| Total Nodes | 99486 |
| :--- | :--- |
| Total Elements | 49183 |
| Maximum Aspect Ratio | 6.2275 |
| \% of elements with Aspect Ratio <3 | 98.1 |
| \% of elements with Aspect Ratio $>\mathbf{1 0}$ | 0 |
| \% of distorted elements(Jacobian) | 0 |

APPENDIX C: Rivet Nuts for Plate Supports
Rivenut-Plusnut


Optimal Configuration


Plusnut
1/4-20-Thead Size (S25P500)
Grip $=1 / 4^{\circ}$
Hole size $=0.347^{\prime \prime}$
Body diameter $=0.346^{\prime \prime}$
Head/Flange height $=0.058^{\prime \prime}$
Head Diameter $=0.625^{\prime \prime}$
Overall length $=1.292^{\prime \prime}$
Final length $=0.505^{\prime \prime}$

## RIVNUT® PN - PLUSNUT® ${ }^{\circledR}$

The Rivnut® PN - Plusnut® has been designed to provide the ultimate pull out strength in thin sheet metals and plastic. Its slotted body splits into four legs providing a wide load-bearing surface on the backside of the parent material. The Rivnut(1) PN - Plusnut@ also features the widest grip range of any blind threaded nut. A grip ID mark is included on the head of the fastener for grip range identification.


Class 2B or Metric Class 6H threads
Steel C1010-1008
Cadmium plating per QQ-P-416
All Dimensions Shown In Inches

| d | e | H |  | D | E | B | L | 12 |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Thread Size | Grip Range | Hole Size |  | Body Diarneter Max. | Head Height Ref. | Head Diameter Ref. | Overall Length Ref. | Installed Length Ret. | Product Code |
|  |  | Min. | Max. |  |  |  |  |  |  |
| 6.32 | 0.020-0.150 | 0.209 | 0.214 | 0.208 | 0.032 | 0.438 | 0.688 | 0.335 | RN632150PN |
| 6-32 | 0.150-0.270 | 0.209 | 0.214 | 0.208 | 0.032 | 0.438 | 0.797 | 0.335 | RNG32270PN |
| 8.32 | 0.020-0.150 | 0.242 | 0.247 | 0.241 | 0.038 | 0.438 | 0.694 | 0.340 | RN832150PN |
| 8.32 | 0.150-0.270 | 0.242 | 0.247 | 0.241 | 0.038 | 0.438 | 0.819 | 0.340 | RN832270PN |
| 10-32 | 0.020-0.175 | 0.273 | 0.278 | 0.272 | 0.038 | 0.500 | 0.819 | 0.425 | RN1032175PN |
| 10-32 | 0.175-0.320 | 0.273 | 0.278 | 0.272 | 0.038 | 0.500 | 0.959 | 0.425 | FN1032320PN |
| 1/4-20 | 0.020-0.280 | 0.347 | 0.352 | 0.346 | 0.058 | 0.625 | 1.058 | 0.505 | RN2520280PN |
| 1/4-20 | 0.280-0.500 | 0.347 | 0.352 | 0.346 | 0.058 | 0.625 | 1.292 | 0.505 | FN2520600PN |
| 5/16-18 | 0.020-0.280 | 0.438 | 0.443 | 0.437 | 0.062 | 0.750 | 1.203 | 0.570 | FN5161828PN |
| 5/16-18 | 0.280-0.500 | 0.438 | 0.443 | 0.437 | 0.062 | 0.750 | 1.437 | 0.570 | FN5161850PN |
| 3/8-16 | 0.020-0.280 | 0.515 | 0.522 | 0.514 | 0.088 | 0.875 | 1.306 | 0.605 | RN3816280PN |
| 3/8-16 | 0.280-0.500 | 0.515 | 0.522 | 0.514 | 0.088 | 0.875 | 1.525 | 0.605 | RN3816500PN |

All Dimensions Shown In Millimeters

| d | e | H |  | D | E | B | L | 12 |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Thread Size | Grip Range | Hole Size |  | $\begin{aligned} & \text { Body } \\ & \text { Diameter } \\ & \text { Max. } \end{aligned}$ | Head Height Ret. | $\begin{gathered} \text { Head } \\ \text { Diameter } \\ \text { Ref. } \end{gathered}$ | Overall Length Ref. | Installed Length Ref. | Product Code |
|  |  | Min. | Max. |  |  |  |  |  |  |
| M4x0.7 | 0.50-3.80 | 6.13 | 6.25 | 6.12 | 0.96 | 11.1 | 17.6 | 8.6 | RN47038PN |
| M4x0.7 | 3.80-6.85 | 6.13 | 6.25 | 6.12 | 0.96 | 11.1 | 20.8 | 8.6 | RN47068PN |
| M5x0.8 | 0.50-4.45 | 7.48 | 7.62 | 7.47 | 0.96 | 12.7 | 22.0 | 9.9 | RN58045PN |
| M5x0.8 | 4.45-8.10 | 7.48 | 7.62 | 7.47 | 0.96 | 12.7 | 24.8 | 9.9 | RN58081PN |
| M6x1.0 | 0.50-7.10 | 8.80 | 8.93 | 8.79 | 1.50 | 15.9 | 26.9 | 12.8 | RN61071PN |
| M6x1.0 | 7.10-12.70 | 8.80 | 8.93 | 8.79 | 1.50 | 15.9 | 32.8 | 12.8 | RN610127PN |
| M8x1. 25 | 0.50-7.10 | 11.11 | 11.50 | 11.10 | 1.57 | 19.0 | 30.5 | 14.5 | RN812571PN |
| M8x 125 | 7.10-12.70 | 11.11 | 11,50 | 11.10 | 1.57 | 19.0 | 36.5 | 14.5 | RN8125127PN |
| M10x1.50 | 0.50-7.10 | 13.07 | 13.26 | 13.06 | 2.24 | 22.2 | 33.2 | 15.8 | RN101571PN |
| M10x1.50 | 7.10-12.70 | 13.07 | 13.26 | 13.06 | 2.24 | 22.2 | 38.7 | 15.8 | RN1015127PN |

## APPENDIX D: Hand Calculations for Shaft and Gate Design

Thesis-HandCalcs Shaft Analysis Larinda Nichols $1 / 4$ Given: 20 psi (air pressure, prototype)

4 psi (ain pressure, 1:5 sectional model)
1,3005 psi (hydrostatic pressure, prototype)
0.2601 psi (hydrostatic pressure, 1:5 sect model)

Dimensions of Prototype:
$1^{\prime} \times 10^{\prime}$ Opening
9 hevels/channels
Each plate $=1 / 4^{\prime \prime}$ thick $\left(9\right.$ gates $\left.\times 1 / 4^{\prime \prime}=2.25^{\prime \prime}\right)$
Remaining Space (Gate) $=117^{\circ} 75^{\prime \prime}$
Each gate $=11.775^{\prime \prime}$ high


Dimensions of 1:5 Sectional Model:

$$
1: 5 \text { for each gate height }=2.355^{\prime \prime}
$$



* Gates should open in $\sim 0.1 \mathrm{sec}$

$$
\begin{array}{ll}
90^{\circ} & \left(\frac{90^{\circ}}{\text { terthsec }}\right)\left(\frac{10 \operatorname{tenth} \sec s}{1 \sec }\right)=\frac{900^{\circ}}{\sec } \\
\frac{900^{\circ}}{\sec }\left(\frac{2 \pi}{360^{\circ}}\right)=15.708 \mathrm{rad} / \mathrm{sec} \\
(\mathrm{\omega e})
\end{array}
$$

Assume standard A36 (ASTM) Steel:
Revolutions: $\frac{1 / 4 \text { revolution }}{0.1 \text { sec }}=\frac{1 \text { revolution }}{0.4 \text { secs }}(2.5)=\frac{2.5 \text { nev }}{\mathrm{sec}}$

$$
\frac{2.5 \mathrm{sec}}{\sec }\left(\frac{60 \mathrm{sec}}{\min }\right)=150 \mathrm{rev} / \min _{(\text {rotation })}(\omega)
$$

Torque:


SRA. L. Nichols

Need to incorporate pressure from pressurized air. Magnitude and Location of Total Force $\begin{aligned} & F_{R_{A}}=\text { Hydrostatic } \\ & F_{R A}=A_{i} P_{r e s s i c e}\end{aligned}$

$$
F_{R_{H}}=\gamma h_{C} A, \quad F_{R_{A}}=P A
$$

For Prototype, (assuming $3^{\prime}$ of total Head)
$F_{R_{n-p}}=\left(62.4 \mathrm{lb} / \mathrm{ft}^{3}\right)(2.5 \mathrm{ft})(1 \mathrm{ft})(10 \mathrm{ft})=1560 \mathrm{lbf}$
$F_{R_{A-P}}=(20 \mathrm{poi})(12 \mathrm{in})(120 \mathrm{in})=28,800 \mathrm{lbf}$
Total $F_{R}=30,360 \mathrm{lbf}$
For 1:5 Scaled Sectional Model, (assuming 7.2" Head)
$F_{R_{H-M}}=\left(62.4 \mathrm{lb} / \mathrm{ft}^{3}\right)(0.5 \mathrm{ft})(0.2 \mathrm{ft})(2 \mathrm{ft})=12.48 \mathrm{lbf}$
$F_{R_{A-M}}=(4 \mathrm{psi})(2.4 \mathrm{in})(24 \mathrm{in})=230.4 \mathrm{lbf} \Rightarrow F_{R}=242.88$
Center of Pressure for Prototype,

$$
x_{R}=\frac{I_{x y c}}{y_{c A}}+x_{c}, \quad y_{R}=\frac{I_{x c}}{y_{c} A}+y_{c}
$$

$$
X_{R}=0 \text { (area is symmetric) }
$$

$$
\begin{aligned}
& y_{R}=\frac{\frac{1}{12}(120 \mathrm{in})(12 \mathrm{in})^{3}}{(30 \mathrm{in})(120 \mathrm{in})(12 \mathrm{in})}+(30 \mathrm{in})= 30.4 \mathrm{in.} \\
& \sim 5.6 \mathrm{in} \text {.above shaft }
\end{aligned}
$$

Center of Pressure for $1: 5$ sectional model,

$$
x_{R}=0
$$

$$
y_{R}=\frac{1 / 12(24 \mathrm{in})(2.4 \mathrm{in})^{3}}{(6 \mathrm{in})(24 \mathrm{in})(2.4 \mathrm{in})}+(6 \mathrm{in})=\underset{\underset{\sim}{6.08 \text { in }}}{\stackrel{1.12 \text { in above shaft }}{ } \quad .}
$$

So, torque on Shaft is:


Prototype $\Rightarrow \tau_{p}=(30,360 \mathrm{lbf})(5.6 \mathrm{in})=170,016 \mathrm{lbf}-\mathrm{in}$ $1: 5$ Sectional $\Rightarrow \tau_{m}=(242.88 \mathrm{lbf})(1.12 \mathrm{in})=272.0256 \mathrm{lbf}-\mathrm{in}$
Model
S.A.
With 2.0 Factor of Safety,

Prototype $F_{R}=60,720 \mathrm{lbf} \quad \tau=340,032 \mathrm{lbf}-\mathrm{in}$. Model $\quad F_{R}=485.76 \mathrm{lbf} \quad \tau=544.0512 \mathrm{lbf}-\mathrm{in}$.
Determine required thickness of Gate,
Treating gate as cantilever beam

$$
\begin{array}{ll}
{ }^{w} & P_{p}=\frac{60,720 \mathrm{lbf}}{(12 \mathrm{in})(12 \mathrm{in})}=42.167 \mathrm{psi} \\
\delta_{\text {max }} & P_{m}=\frac{485.76 \mathrm{lbf}}{(2.4 \mathrm{~min})(24 \mathrm{in})}=8.4333 \mathrm{psi}
\end{array}
$$

$$
w_{p}=(42.167 \mathrm{psi})(120 \mathrm{in})=5060.04 \mathrm{lff} / \mathrm{in}
$$

$$
W_{m}=(8.4333 \text { pis })(24 \mathrm{in})=202.399 \mathrm{lbf} / \mathrm{in}
$$

For $1 / 16^{\text {" }}$ maximum deflection,

$$
\begin{aligned}
& \delta_{\text {max }}=-\frac{w L^{4}}{8 E 1} \rightarrow \delta_{\text {max }}=\frac{-w L^{4}}{8 E\left(1 / b h^{3}\right)} \rightarrow h^{3}=\frac{-w L^{4}}{(1 / 16 i n) 8 E(/ 12 b)} \\
& h^{3} p=-\frac{(5060.04 \mathrm{lbf} / \mathrm{in})(12 \mathrm{in})^{4}}{(1 / 6 \mathrm{in}) 8\left(29 \times 10^{6} p \mathrm{si}\right)(1 / 2) 120 \mathrm{in}}=0.7236 \mathrm{in}^{3}
\end{aligned}
$$

minimum thickness, $h_{p}=0.8978$ in

$$
h_{m}^{3}=\frac{-(202.3991 \mathrm{bf} / \mathrm{in})(2.4 \mathrm{in})^{4}}{(1 / 16 \mathrm{in}) 8\left(29 \times 10^{2} \mathrm{psi}\right)(1 / 12) 24 \mathrm{in}}=0.000232 \mathrm{in}^{3}
$$

$$
\text { minimum thickness, } h_{m}=0.0614 \text { in }
$$

Confirming results with solidworks FEA using $h_{p}$ क $h_{m}$,

$$
\begin{aligned}
& f_{P}=0.059 \mathrm{in} \\
& \delta_{m}=0.059 \mathrm{in}
\end{aligned}
$$

Standard Plate thicknesses will be chosen $h_{p}=1 \mathrm{in}$.

$$
h_{m}=3 / 16 \mathrm{in} .
$$

$$
(0.0125 \text { in less than } 1 / 5 \text {, but closest found) }
$$

SA.
So with 2.0 FOS, gate thickness results in Solidworks FEA,
Prototype, $h_{p}=1.0 \mathrm{in}, \delta_{p}=0.043 \mathrm{in}, \sigma_{p}=16,489 \mathrm{psi}$
Model, $h_{m}=316 \mathrm{in}, \delta_{m}=0.002 \mathrm{in}, \sigma_{m}=3.767 \mathrm{psi}$
Next, determining shaft diameter. Treating gate as "Key",
Torsion: $\tau_{\text {max }}=\frac{T_{r}}{J}$, where $J=\frac{\pi d^{4}}{32}$
Then, $\tau_{\text {max }}=\frac{16 T}{\pi d^{3}}$
Maximum Bending Stress: $\sigma_{\max }=\frac{32 M}{\pi d^{3}}$
For Prototype, assume $1.25^{\prime \prime}$ diameter rod

$$
\begin{aligned}
& \tau_{\max }=\frac{16(340,032 \mathrm{lb}-\mathrm{in})}{\pi(1.25 \mathrm{in})^{3}}=886,666 \text { psi }
\end{aligned}
$$

Rod Calculations


A Rod must be designed to withotand the anticipated bending + torsion due to the pressure force on the attached gate.
Plate thickness $=3 / 8$ in. Gate thickness $=T B D$
Rod Length $=10 \mathrm{ft}+$ Sides Pressure $=1.3$ psi (Hydro) +20 psi (air)

Bending+Torsion
From Shigley's: $\sigma_{a}=\frac{K_{f} M a C}{I}, \sigma_{m}=\frac{K_{f} M_{m C}}{I}, \tau_{\text {max }}=\frac{16 T}{\pi d^{3}}$
For $P_{A}=20.0 \mathrm{psi}$, width of gate $=10 \mathrm{ft}$, height of gate $=12 \mathrm{in}$.


$$
\begin{aligned}
& q=20.3 p s i, \omega=120 \mathrm{in}, H=12 \mathrm{in} . \\
& M_{A I R}=q \omega x d x \\
& M_{\text {AIR }}=\int_{0}^{H} q \omega x d x \\
& M_{\text {AIR }}=q \omega \int_{0}^{H} x d x=1 /\left.2 q \omega x^{2}\right|_{0} ^{H} \\
& M_{\text {AIR }}=\left.\frac{\left(201 \mathrm{~b} / \mathrm{in}^{2}\right)(120 \mathrm{in}) x^{2}}{2}\right|_{0} ^{12} \\
& M_{\text {AIR }}=172,800 \mathrm{lb}-\mathrm{in}
\end{aligned}
$$

For $P_{\mathrm{H}_{2} \mathrm{O}}=1.3$ psi, $q=1.3 \mathrm{psi}, \omega=120 \mathrm{in}, H=36 \mathrm{in}$.

$P=\gamma h$ and $F=P A$,

$$
\begin{aligned}
& P=\gamma(H-x) \\
& A=\omega \theta x \\
& F=\gamma \omega(H-x) d x \\
& M_{H_{2} O}=\gamma \omega x(H-x) d x \\
& M_{H_{2} O}=\int_{0}^{H} \gamma \omega x(H-x) d x
\end{aligned}
$$

Continued $\rightarrow$

RC.

$$
\begin{aligned}
& M_{\mathrm{H}_{20}}=\gamma \omega \int_{0}^{H} H x-\int_{0}^{H} x^{2} d x \\
& M_{\mathrm{H}_{2} \mathrm{O}}=\gamma \omega\left[\left.\frac{\mathrm{H} x^{2}}{2}\right|_{0} ^{H}-\left.\frac{x^{3}}{3}\right|_{0} ^{H}\right], H=3 \mathrm{ft}, \omega=10 \mathrm{ft} \\
& M_{\mathrm{H}_{2} \mathrm{O}}=\left(62.4 \mathrm{lb} / \mathrm{ft}^{3}\right)(10 \mathrm{ft})\left[\frac{27 \mathrm{ft}^{3}}{2}-\frac{27 \mathrm{ft}^{3}}{3}\right] \\
& M_{\mathrm{H}_{20} \mathrm{O}}=2808 \mathrm{lbf}-\mathrm{ft}=33,696 \mathrm{lbf}-\mathrm{in} .
\end{aligned}
$$

Total Bending moment acting on Rod,

$$
\begin{aligned}
& M_{\text {ARR }}+M_{\mathrm{H}_{2} \mathrm{O}}=(172,800 \mathrm{lb}-\mathrm{in}+33,696 \mathrm{lb} \text {-in })=206,496 \mathrm{lb}-\mathrm{in} \\
& \sigma=\frac{32 M_{a}}{\pi d^{3}} \rightarrow d=\left(\frac{32 M_{a}}{\pi \sigma}\right)^{1 / 3}
\end{aligned}
$$

yield stress of $A 36$ steel, $\sigma_{y}=36,000$ psi

$$
d=\left(\frac{32(206,496 \mathrm{lb-in})}{\pi\left(36,000 \mathrm{lb} / \mathrm{in}^{2}\right)}\right)^{1 / 3}=3.88^{\prime \prime}
$$

So, a $3.88^{\prime \prime}$ dia. A36 rod will yield under this load
We can't have a $4^{\prime \prime}$ dian. nod, so the load must be decreased substantially. If we can wait to apply the 20 psi of pressure at the moment we wien the gates, we the decrease the moment to 33,6 ? 16 pion in for only the 3 'of static head.

$$
d=\left(\frac{32(33,696 \mathrm{lb}-\mathrm{in})}{\pi\left(36,000 \mathrm{lb} / \mathrm{in}^{2}\right)}\right)^{1 / 3}=2.12^{\prime \prime}
$$

This is still too large and at yield. This nod would take up $y_{6}$ of our orifice.
If we decrease the static head to only account for the water in the channel (lIft),

$$
\begin{aligned}
& M=\int_{0}^{1} \gamma \omega x(h-x) d x=\gamma \omega\left[\left.\frac{H x^{2}}{2}\right|_{0} ^{1}-\left.\frac{x^{3}}{3}\right|_{0} ^{1}\right] \\
& M=\left(62.4 \mathrm{lb} / \mathrm{ft}^{3}\right)(10 \mathrm{ft})\left[\frac{1}{2}-\frac{1}{3}\right] \mathrm{ft}^{3}=1041 \mathrm{lb}-\mathrm{ft}=1248 \mathrm{lb-in} \\
& d=\left(\frac{32(1248 \mathrm{lb}-\mathrm{in})}{\pi\left(36,000 \mathrm{lb} / \mathrm{in}^{2}\right)}\right)^{1 / 3}=0.7068^{\prime \prime} \quad \text { (acceptable) }
\end{aligned}
$$

RC.
Now we need to look at the gate thickness.
As a Cantilever, $l=12^{\prime \prime}, b=120^{\prime \prime}, h=$ TAD


$$
\begin{aligned}
\sigma_{B} & =\frac{M_{C}}{I}, C=\frac{h}{2} \text { and } I=\frac{b h^{3}}{12} \\
& \rightarrow \frac{I}{c}=\frac{M}{\sigma} \rightarrow \frac{b h^{32} 2}{h 12}=\frac{M}{\sigma}
\end{aligned}
$$

Finally, $h=\left(\frac{12 M}{2 b \sigma}\right)^{1 / 2}$
For our $M=1248 \mathrm{lb}$-in,

$$
h=\left(\frac{12(12481 \mathrm{bin})}{2(120 \mathrm{in})\left(36,000 \mathrm{ib} / \mathrm{in}^{2}\right)}\right)^{1 / 2}=0.0416^{\prime \prime}
$$

Now to check this deflection, $F=\frac{M}{x_{c}}=\frac{12481 \mathrm{~b}-\mathrm{in}}{2}=6241 \mathrm{bf}$

$$
\begin{aligned}
& y_{\text {max }}=\frac{F l^{3}}{3 E I}=\frac{(624 \mathrm{lbf})(12 \mathrm{in})^{3}}{3(29 E 6 \mathrm{psin})\left(\mathrm{y}_{12}(20 . \mathrm{n})(0.046 \mathrm{in})^{3}\right.} \\
& y_{\text {max }}\left(h=0.0416^{\prime \prime}\right)=17.22 \mathrm{in.} \text { (Not acceptable) }
\end{aligned}
$$

Working backward,

$$
\begin{aligned}
& y_{\text {max }}=\frac{F l^{3}}{3 E I}, \text { set } y_{\text {max }}=0.01 \text { in. } \\
& I=\frac{F l^{3}}{3 E y} \rightarrow \frac{b h^{3}}{12}=\frac{F l^{3}}{3 E y} \rightarrow h=\left(\frac{12 F l^{3}}{3 b E y}\right)^{1 / 3} \\
& h=\left(\frac{12(62416 f)(12 \mathrm{in})^{3}}{3(120 \mathrm{im})(29 E 6 p \mathrm{pi})(0.01 \mathrm{in})}\right)^{1 / 3}=0.499 \mathrm{in} .
\end{aligned}
$$

So the gate will have to be $1 / 2^{\prime \prime}$ thick.
From Grainger. com
Keyed Shaft for $1 / 2$ in Key size $=2^{\prime \prime}$


RC.
L. Nichols 4/4

This is larger than we had hoped, but it is the best option so far.
Results and support options will be checked using
Solidworks.

$$
\begin{aligned}
& \text { For } 2^{\prime \prime} \text { Shaft, } M=1248 / \mathrm{b}-\mathrm{in} \\
& \sigma=\frac{32 M}{\pi d^{3}} \\
& 2^{1 / 2} \\
& \sigma=\left(\frac{32\left(1248 / 4^{\prime \prime}-i n\right)}{\pi(2 \mathrm{in})^{3}}\right)=1589 \mathrm{pai} \\
& \quad \text { EOS }=22.7
\end{aligned}
$$

Options for hinges, supports need to be explored to se if a smaller Shaft might be suitable with
For Torsion: $\tau_{\max }=\frac{16 T}{\pi d^{3}}=\left(\frac{16(24961 \mathrm{~b}-\mathrm{in})}{\pi(2: i n)^{3}}\right)=1589: 0 \mathrm{psi}$

* To Discuss, how much deflection can be allowed In gates to avoid overclesign of shaft. (Shaft is mote by Also is it feasible to decrease $P=0.433$ for only lIft static head, no 3 ft head, no air pressure?

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## KZYZD - IMPERIAL

## DESCRIPTION

Fully keyed shafts provide a good fit between the key and the key way. Key ways are cut in accordance with ANSI B17.1. Partial key ways and other materials are available including turned and polished and medium carbon steel (suitable to heat treat).

## HOW TO IDENTIFY

1. Measure diameter ( $A$ ).
2. Measure length (C)
3. Build the part number from the chart on the next page.


## PREFIX MATERIAL/FINISH

$K F=$ COLD FINISHED STEEL, PLAIN* (1Ө18)
KF44 = COLD FINISHED STEEL, PLAIN* (1144)
KF4S = COLD FINISHED STEEL, PLAIN* ( $1 \ominus 45$ )
$K F 96=316$ STAINLESS STEEL, FLAIN
KFTP = COLD FINISHED STEEL, PLAIN* (1Ө45)
TURNED \& POLISHED
$K F Z \quad$ - COLD FINISHED STEEL, ZINC CLEAR (1Ө18)

| DIAMETER (A) TOLERANCES |  |  |  |
| :---: | :---: | :---: | :---: |
| MATER\|AL |  | DIAMETER |  |
| (Prefto) | (Material/Finish) | (Slze Range) | (Tolerance) |
| KF | Cold Finished Steel, Plain* (1018) | All Sizes | +06.0020* |
| KF44 | Cold Finished Steel, Plain* (1144) |  |  |
| KF45 | Cold Finished Steel, Plain* (1045) | All Sizes | + $\alpha=.0010^{*}$ |
| KFS | 303 stainless steel, Plain |  |  |
| KFS6 | 316 Stainless Steel, Plain | All Sizes | +0\%.0020* |
| KFTP | Cold Finished Steel, Plain* (1045), <br> Turned \& Polished | All Sizes | -.0005/-.0015* |
| KFZ | Cold FInished steel, Zinc Clear (1018) | All Sizes | +01-.0020* |

## KEYED SHAFTING CAN BE

 PRODUCED IN MANT DIAMETER AND LENGTH COMBINATIONS| KEY UAY (D $\times$ B) SIZE |  |
| :---: | :---: |
| DIAMETER | KEr WAr |
| (shaft size Range) | (D) $\times(B)$ |
| $1 / 2^{*}$ | $1 / 8^{*} \times 1 / 16^{*}$ |
| 5/8-7/8* | $3 / 16^{*} \times 3 / 32^{*}$ |
| 15/16-1-1/4* | $1 / 4^{*} \times 1 / 8^{*}$ |
| 1-5/16-1-3/8* | $5 / 16^{*} \times 5 / 32^{*}$ |
| 1-7/16-1-3/4* | $3 / 88^{*} \times 3 / 16^{*}$ |
| 1-7/8-2-3/16 ${ }^{\text { }}$ | $1 / 2^{*} \times 1 / 4^{*}$ |
| $2-7 / 16-2-1 / 2^{*}$ | $5 / 88^{*} \times 5 / 16^{\circ}$ |
| 2-15/16-3-1/4* | $3 / 4^{*} \times 3 / 8{ }^{*}$ |
| 3-3/8-3-3/4* | $7 / 8^{*} \times 7 / 16^{*}$ |
| 3-15/16-4-1/2* | $1 * \times 1 / 2^{*}$ |


| LENCTH (C) TOLERANCES |  |  |
| :--- | :--- | :--- |
| LENGTH |  | TOLERANCE |
| (Suffor) (Length) | (Tolerance) |  |
| -12 | $\left.12^{*}(1)^{\prime}\right)$ | $\pm 2^{*}$ |
| -36 | $36^{\circ}(3)$ |  |
| -72 | $72^{*}\left(6^{\prime}\right)$ |  |
| -144 | $\left.144^{*}(12)^{\prime}\right)$ |  |

Nonstandard lengths up to $144^{*}$ are available. Lengths over $3^{\prime}$ may be subject to a packaging charge.


[^0]
## APPENDIX E: 650 LB Holding Force Electromagnets

## Electromagnetic Locks



2511

## 2500 Series 650 Lb . "Mini"

Mini electromagnetic locks, surface mounted for single and double outswing or inswing doors.
Size: $1-1 / 4^{\prime \prime} \mathrm{D} \times 1-3 / 4^{\prime \prime} \mathrm{H} \times 10-1 / 2^{\prime \prime} \mathrm{L}$ Single or $21^{\prime \prime} \mathrm{L}$ Double
Multi-voltage field selectable $12 / 24$ VAC/VDC
Current Draw: 500 mA @ 12 V Single / 1 Amp Double
$250 \mathrm{~mA} @ 24 \mathrm{~V}$ Single / 0.50 Amps Double

| MODEL | PRODUCT | FINISH | WEIGHT |
| :--- | :--- | :---: | :---: |
| 2511 | Single Electromagnetic Lock Outswing | US28 | 4.5 Lbs. |
| 2522 | Double Electromagnetic Lock Pair Outswing | US28 | 9 Lbs. |
| 25117 | Single Electromagnetic Lock Inswing | US28 | 7.5 Lbs. |
| 2522T32 | Double Electromagnetic Lock Pair Inswing | US28 | 15 Lbs. |

## 2585 Series 650 Lb. "Bantam"

"Bantam" electromagnetic lock, surface mounted for single outswing or inswing doors.
Size: $1-1 / 4^{\prime \prime} \mathrm{D} \times 1-3 / 4^{\prime \prime} \mathrm{H} \times 8-1 / 2^{\prime \prime} \mathrm{L}$
Field Selectable $12 / 24$ VDC.
Current Draw: 500 mA @ 12VDC
250 mA @ 24VDC



2585-TJ85

OPTIONS

| - ATS | Anti-Tamper Switch - Signals removal of the housing cover. |
| :---: | :---: |
| - CLH | Custom Length Housing - See page 24. |
| DSM | Door Status Switch - Signals door closed or ajar. |
| DSM2 | Door Status Switch - For the 2522 Series only. |
| DYN | Dynastat Force Sensor - Indicates efficient magnetic bond. |
| DYN2 | Dynastat Force Sensor - For the 2522 Series only. |
| LED | Bi-Color LED - For local signaling of lock status. (Requires DYN Option. Not available on TJ models.) |
| LED2 | Bi-Color LED's (2) - For the 2522 Series only. (Requires DYN2 Option. Not available on TJ models.) |
| SF | Special Finish - Standard finish US28. Special anodized finishes see page 56. |

Notes: - ATS and CLH options not available on the 2585 Series.

Note: For AC operation order \#7088 Pre-wired Rectifier Bridge. See page 48.

APPENDIX F: Hand Calculations for Angular Velocity of Shaft

SHAFT - $\omega$
L. Nichols

Angular velocity of shaft from 0.433 psi


Moment from $0.433 \mathrm{psi}=1248 \mathrm{lb}-\mathrm{in}$

$$
\begin{aligned}
& \theta=90^{\circ}=\pi / 2 \text { radians } \\
& T=(1248 \mathrm{lb} \text {-in })(\pi / 2)=1960 \mathrm{lb} \text {-in }
\end{aligned}
$$

Since we accelerating from rest, the moment right before we finish rotating $=t_{2} \quad\left(t_{1}=0\right)$
Using Work-Energy Principle,

$$
\begin{aligned}
& T_{1}+V_{1}+\left(U_{1} /_{2}\right)_{n c}^{\text {ext }}+\left(U_{1-2}\right)_{n c}^{i n t}=T_{2}+V_{2} \\
& \left(U_{1-2}\right)_{n c}^{\text {int }}=T_{2}=\frac{1}{2} m v_{2}^{2}+2 I_{G} \omega_{w 2}^{2} \\
& m_{\text {ass }}(m) \text { of } 10 \mathrm{~d} \approx 60 \mathrm{lbm} \\
& I_{G}=\frac{m r^{2}}{2}=\frac{60 \mathrm{lbm}\left(\frac{1.7 \mathrm{sin}}{2}\right)^{2}=23 \mathrm{lbm}-\mathrm{in}^{2}}{2}
\end{aligned}
$$

Need $w_{1,}$ and $v_{2}$.
Since $\omega_{a n}=\frac{v_{2}}{r} \rightarrow T_{2}=\frac{1}{2} m v_{2}^{2}+2 I_{G}\left(\frac{v_{2}}{r}\right)^{2}$

$$
\begin{aligned}
& T_{2}=1 / 2 m v_{2}^{2}+2 I_{G}\left(v_{2}^{2} / r^{2}\right) \\
& T_{2}=v_{2}^{2}\left(1 / 2 m+2 / r^{2} I_{G}\right) \\
& v_{2}=\left(\frac{T_{2}}{1 / 2 m+2 / r^{2} I_{G}}\right)^{1 / 2}
\end{aligned}
$$

$\left.v_{2}=\left(\frac{1960 \mathrm{lbf}-\mathrm{in}}{601 \mathrm{bm} / 2+2 / 0.765625 \mathrm{in}^{2}\left(231 \mathrm{bm}-\mathrm{in}^{2}\right)}\right)\left(\frac{1 \mathrm{bm}-32.2 \mathrm{ff}}{16 f-\mathrm{s}^{2}}\right)\left(\frac{12 \mathrm{in}}{1 \mathrm{fE}}\right)\right)^{1 / 2}$
$v_{2}=79.4 \mathrm{in} /$ second

$$
\omega_{\omega_{2}}=\frac{v_{2}}{r}=\frac{79.4 \mathrm{in} / \mathrm{s}}{0.875 \mathrm{in}}=90.76 \mathrm{rad} / \mathrm{s}
$$

$t=\pi / 2 \mathrm{rad}\left(\frac{s}{90.76 \mathrm{rad}}\right)=\underline{\underline{0.0173 ~ s e c o n d s}}$

## APPENDIX G: Pneumatics

Thesis Pressure Calculations Larinda Nichols 1/8
Dynamics - Rectilinear Motion for Prototype to Verify Method
Per Level,


Where $l=$ length of water column,$d=$ wave development length

* Neglecting Friction in all calculations

Known Velocity $(V)=25.4 \mathrm{ft} / \mathrm{s}$
Prototype level height $(h)=1 \mathrm{ft} /$ level
Prototype level width $(\omega)=10 \mathrm{ft}$
Dynamics Equations for Acceleration (a), Velocity $(v)$,

$$
\begin{aligned}
& a=\frac{d v}{d t}, a=\beta t^{2} \text { (where } \beta \text { is some unknown constant) } \\
& \text { so, } \beta t^{2}=\frac{d v}{d t} \rightarrow d v=\beta t^{2} \rightarrow \int_{0}^{v} d v=\int_{0}^{t} \beta t^{2} d t \rightarrow V=1 / 3 \beta t^{3} \\
& V=\frac{d s}{d t} \rightarrow 1 / 3 \beta t^{3}=\frac{d s}{d t} \rightarrow d s=1 / 3 \beta t^{3} d t \rightarrow \int_{s_{0}}^{5} d s=\int_{0}^{t / 3 \beta} \beta t^{3}
\end{aligned}
$$

$$
\text { When } t=t_{f}, V=v_{f} \text {, and } s-S_{0}=d=1 / 12 \beta t_{t}^{4}
$$

$$
\text { Then } V_{f}=1 / 3 \beta t_{f}{ }^{3}
$$

$$
\text { So, } \beta=\frac{3 V_{f}^{4}}{64 d^{3}} \text { and } t_{f}=\frac{4 d}{V_{f}} \quad \text { [Ref: Engineering Mechanics } \frac{\text { Dynamics -Gray, Ind } d \text { ] }}{}
$$

When $V_{f}=25.4 \mathrm{ft} / \mathrm{s}$ and $d=15 \mathrm{ft}$

$$
\beta=\frac{3(25.4 \mathrm{ft} / \mathrm{s})^{4}}{64(15 \mathrm{ft})^{3}}=5.78 \mathrm{ft} / \mathrm{s}^{4} \quad t_{f}=\frac{4(15 \mathrm{ft})}{25.4 \mathrm{ft/s}}=2.36 \mathrm{~s}
$$

Then, $a=\beta t^{2}=\left(5.78 \mathrm{ft} / \mathrm{s}^{4}\right)(2.36 \mathrm{~s})^{2}=32.3 \mathrm{ft} / \mathrm{s}^{2}$
If $l=15 \mathrm{ft}, \forall=(1 \mathrm{ft})(10 \mathrm{ft})(15 \mathrm{ft})=150 \mathrm{ft}^{3}$ with $A=10 \mathrm{ft}^{2}$
$m=\rho \forall=\left(62.4 \mathrm{lbm} / \mathrm{ft}^{3}\right)\left(150 \mathrm{ft}^{3}\right)=9360 \mathrm{lbm}$
$F=m a=(9360 \mathrm{lbm})\left(\frac{\mathrm{bt}-\mathrm{s}^{2}}{32.21 \mathrm{lmm-ft}}\right)\left(32.3 \mathrm{ft} / \mathrm{s}^{2}\right)=9376.9 \mathrm{lbf}$

$P C$.
If $d$ is decreased to only 5 ft develop distance,

$$
\begin{aligned}
& \beta=\frac{3\left(25.4 \mathrm{ff}(\mathrm{~s})^{4}\right.}{64(5 \mathrm{ft})^{3}}=156 \mathrm{ft} / \mathrm{s}^{4} \quad t_{f}=\frac{4(5 \mathrm{ft})}{25.4 \mathrm{ft} / \mathrm{s}}=0.787 \mathrm{~s} \\
& a=\left(156 \mathrm{ft} / \mathrm{s}^{4}\right)(0.787 \mathrm{~s})^{2}=96.77 \mathrm{ft} / \mathrm{s}^{2} \\
& m=\left(62.4 \mathrm{lbm} / \mathrm{ft}^{3}\right)\left(150 \mathrm{ft}^{3}\right)=9360 \mathrm{lbm} \\
& F=(9360 \mathrm{lbm})\left(96.77 \mathrm{ft} / \mathrm{s}^{2}\right)\left(16 \mathrm{f}-\mathrm{s}^{2} / 1 \mathrm{lbm} \cdot 32.2 \mathrm{ft}\right)=28,131 \mathrm{lbf} \\
& P=\frac{28.131 \mathrm{lbf}}{16 \mathrm{ft}^{2}\left(144 \mathrm{in}^{2}\left(\mathrm{ft}^{2}\right)\right.}=19.5 \text { poi Required to accelerate } \\
& (1 \times 10 \times 15) \mathrm{ft} \text { level to } 25.4 \mathrm{f} / \mathrm{s} \\
& \text { in } 5 \mathrm{ft} \text {, per level, in } 0.787 \mathrm{~s} \text {. } \\
& \text { (Close agreement with Fow-30 V) }
\end{aligned}
$$

For Pipe Model,
$\begin{array}{ll}\text { Schedule } & 4 N P S \rightarrow 0 . D=4.5 \mathrm{in}, I . D=4.026 \mathrm{in}, A=12.73 \mathrm{in}^{2} \\ 40 P V C & Y_{2} N P S \rightarrow 0.0 .=0.840, I . D=0.622 \mathrm{in}, A=0.304 \mathrm{~m}^{2}\end{array}$

$$
V_{f}=(\text { scaled from Prototype })=11.36 \mathrm{ft} / \mathrm{s} \quad\left[R_{0 j i n} T .\right]
$$



Overall Run length of $(d)=5 \mathrm{ft}+(2 \mathrm{ft}-1.41 \mathrm{ft}) \approx 5.6 \mathrm{ft}$
Overall Volume of water in pipe,

$$
\begin{aligned}
& (l)=1.41 \mathrm{ft}+2 \mathrm{ft}+1.41 \mathrm{ft}=4.82 \mathrm{ft} \\
& (r)=2.013 \mathrm{in}=0.16775 \mathrm{ft} \\
& \forall=\ell \mathrm{r}^{2}=(4.82 \mathrm{ft}) \pi(0.16775 \mathrm{ft})^{2}=0.426 \mathrm{ft}^{3}
\end{aligned}
$$

Acceleration from $V=0$ to $V=11.36 \mathrm{ft} / \mathrm{s}$ in 5.6 ft ,

$$
\begin{aligned}
& \beta=\frac{3(11.36 \mathrm{ft} / \mathrm{s})^{4}}{64(5.6 \mathrm{ft})^{3}}=4.45 \mathrm{ft} / \mathrm{s}^{4} \\
& t=\frac{4(5.6 \mathrm{ft})}{(11.36 \mathrm{ft} / \mathrm{s})}=1.97 \mathrm{~s} \\
& a=\left(4.45 \mathrm{ft} / \mathrm{s}^{4}\right)(1.97 \mathrm{~s})^{2}=17.3 \mathrm{ft} / \mathrm{s}^{2} \text { for constant } \\
& \text { acceleration }
\end{aligned}
$$

PC.
Looking at Force required at $t_{0}$
At $t_{0}$,

$\sum F_{x}: F_{A} \cos \theta+N_{0} \cos \theta-N_{(3)} \cos \theta=$ max
$\sum F_{y}:-F_{A} \sin \theta+N_{(1)} \sin \theta-W_{(1)}+N_{(2)}-W_{\theta}+N_{(3)} \sin \theta=m a_{y}$ Also, $W_{1}=W_{3}$, so $N_{1}=N_{3}$

$$
N_{2}=W_{2} \text {, assume } N_{2} \text { curl cancel } W_{2} \text {, so } N_{1}=N_{3}=N
$$

So, $F_{A} \cos \theta+N \cos \theta-N \cos \theta=\max _{x}$

$$
F_{A} \cos \theta=m a_{x}
$$

$$
-F_{A} \sin \theta+2 N \sin \theta-W_{1}-W_{3}=m a_{y}
$$

$$
F_{A} \sin \theta=2 N \sin \theta-W_{1}-W_{3}-m_{a y}
$$

Since $a=a_{x}, a_{y}=0$

$$
F_{A} \sin \theta=2 N \sin \theta-W_{1}-W_{3}
$$

$$
F_{A}=\frac{m a_{x}}{\cos \theta}=\frac{m\left(17.3 \mathrm{ft}^{2}\right)}{0.707}=24.47 \mathrm{~m}
$$

Mass of all 3 pieces.

$$
\forall=\left\{2\left[(16.92 \mathrm{in}) \pi(2.013 \mathrm{in})^{2}\right]+(24 \mathrm{in}) \pi(2.013 \mathrm{in})^{2}\right\}
$$

$$
\forall=736.32 \mathrm{in}^{3}=0.426 \mathrm{ft}^{3}
$$

$$
m=p \forall=\left(62.4 \mathrm{lb} / \mathrm{ft}^{3}\right)\left(0.426 \mathrm{ft}^{3}\right)=26.6 \mathrm{lbm}
$$

$$
F_{A}=\frac{(26.61 \mathrm{bm})\left(16 \mathrm{ff} \cdot \mathrm{~s}^{2} / 32.2 \mathrm{ft}\right)\left(17.3 \mathrm{ft} / \mathrm{s}^{2}\right)}{0.707}=20.21 \mathrm{lbf}
$$

$$
P=\frac{F}{A}=\frac{20.211 \mathrm{lbf}}{12.73 \mathrm{in}^{2}}=1.59 \mathrm{psi}
$$

PC. Lainda Nichols 4/8
Now Breaking down into times to find max $F$ required
At $t_{0}, F_{A} \downarrow \downarrow w_{1}$

$$
F_{W_{2,3}}=W_{2}, N_{2}, W_{3}, N_{3}
$$



$$
\begin{aligned}
\sum F_{x}: & F_{A}+W_{1}+W_{2} \operatorname{Cos} \theta-N_{2} \cos \theta+W_{3} \operatorname{Cos} \theta-N_{3}=m a_{x} \\
F_{A} & =m_{2} g \cos \theta+m_{3} g-W_{1}-W_{2} \cos \theta-W_{3} \cos \theta+m a_{x}
\end{aligned}
$$

$$
F_{A}=m_{2} g \cos \theta+m_{3} g-m_{1} g-m_{2} g \cos \theta-m_{3} g \cos \theta+m_{T} a_{x}
$$

$$
F_{A}=\left(32.2 \mathrm{ft} / \mathrm{s}^{2}\right)[(11 \mathrm{lbm})(\cos \theta)+(1.7816 \mathrm{~km})-(7.781 \mathrm{bm})-(11 \mathrm{lbm})(\cos \theta)
$$

$$
-(7.78 \mathrm{ibm})(\cos \theta)]+(26.6 \mathrm{lbm})\left(17.3 \mathrm{f} 4 \mathrm{~s}^{2}\right)
$$

$$
\begin{aligned}
& F_{A}=\left(283.04 \mathrm{lbm} \mathrm{ft} / \mathrm{s}^{2}\right)\left(1 \mathrm{bf}-\mathrm{s}^{2} / 32.2 \mathrm{lbmft}\right) \\
& F_{A}=8.79 \mathrm{lbf} \quad P=0.69 \mathrm{ps}
\end{aligned}
$$

This pressure would only be valid for a short time until force vector changes, So looking instead at a "Worse Case Scenario"
 ConstantUphill@45 ${ }^{\circ}$


$$
\begin{aligned}
& W_{T}=m_{T}\left(32.2 \mathrm{ft} / \mathrm{s}^{2}\right) \\
& m_{T}=26.6 \mathrm{lbm}
\end{aligned}
$$

$$
\begin{aligned}
& \sum F_{X}: F_{A}-W_{T} \cos \theta=m a \\
& F_{A}=W_{T} \cos \theta+m a \\
& F_{A}=(2661 \mathrm{~lm})\left(32.2 \mathrm{ft} / \mathrm{s}^{2}\right) \cos 45^{\circ}+(26.61 \mathrm{lbm})\left(17.3 \mathrm{ft} / \mathrm{s}^{2}\right) \\
& F_{A}=\left(1065.83 \mathrm{lbm} \mathrm{ft} / \mathrm{s}^{2}\right)\left(1 \mathrm{bf}-\mathrm{s}^{2} / 32.2 \mathrm{lbm} \mathrm{ft}\right) \\
& F_{A}= 33.1 \mathrm{lbf}
\end{aligned}
$$

$P=2.6$ psi to push all water up the $45^{\circ}$ incline

PC Larinda Nichols 5/8
Using $P_{\text {max }}=P=2.6$ and treating air as ideal gas,

$$
\begin{aligned}
& \rho=\frac{P}{R T}=\frac{\left(2.6 \mathrm{lb} / \mathrm{in}^{2}+14.7 \mathrm{lb} / \mathrm{in}^{2}\right)\left(144 \mathrm{in}^{2} / \mathrm{ft}^{2}\right)}{(1716 \mathrm{ftlb} / \operatorname{sig} R)(519 R)}=\frac{0.0028 \mathrm{lb} / \mathrm{ft}^{2}}{\mathrm{ftlb} / \mathrm{slog}} \\
& \rho=\frac{0.0028 \mathrm{lbf} \cdot \operatorname{slog}}{\mathrm{ft} 3 \mathrm{lbf}}\left(\frac{1 \mathrm{bf}-\mathrm{s}^{2}}{\operatorname{slog} \mathrm{ft}}\right)\left(\frac{32.2 \mathrm{ft} \cdot \mathrm{lbm}}{1 \mathrm{bf}-\mathrm{s}^{2}}\right)=\frac{0.09 \mathrm{lbm}}{\mathrm{ft}}=\frac{5.2 \times 10^{-5} \mathrm{lbm}}{\mathrm{in}^{3}}
\end{aligned}
$$

So, with gauge pressure of $2.6 \mathrm{lb} / \mathrm{in}^{2}, \rho_{\text {air }}\left(59^{\circ} \mathrm{F}\right)=\frac{5.2 \times 10^{-5} \mathrm{lbm}}{\mathrm{in}^{3}}$
From Table 1.7, Fundamentals of Fluid Mechanics, Munson, th

$$
\operatorname{Pain}\left(59^{\circ}\right)=2.38 \times 10^{-3} \text { slugs } / \mathrm{ft}^{3}
$$

our case, $\rho=2.8 \times 10^{-3}$ slug $/ \mathrm{ft}^{3}$
Compression ratio is (1.175:1) Not Much
Using continuity equation,

$$
\sum \dot{m}_{\text {out }}=\sum \dot{m} \text { in, where } \dot{m}=\text { mass flow rate }=\rho Q=\rho A V
$$

Then, $\rho A V_{1}=\rho_{2} A_{2} V_{2}$
Since we want constant velocity, $V_{1}=V_{2}$, and

$$
\begin{aligned}
\rho_{1} A_{1}=\rho_{2} A_{2} & \text { where } \begin{aligned}
\rho_{1} & =\text { our calculated density } \\
A_{1} & =\text { flowarea of } 4^{\prime \prime} \mathrm{NPS} P V C \\
A_{1} & =12.73 \mathrm{in}^{2}
\end{aligned} \quad \begin{array}{l}
A_{2}=\text { flow area of } 1 / 2^{2} \mathrm{NPSPVC} \\
A_{2}
\end{array}=0.304 \mathrm{in}^{2} \\
\text { So, } \rho_{2} & =\frac{\rho_{1} A_{1}}{A_{2}}=\frac{\left(2.8 \times 10^{-3} \text { slog } / \mathrm{ft}^{3}\right)\left(\mathrm{ft}^{3} / 1728 \mathrm{in}^{3}\right)\left(12.73 \mathrm{in}^{2}\right)}{\left(0.304 \mathrm{in}^{2}\right)} \\
\rho_{2} & =6.785 \times 10^{-5} \text { slugs } / \mathrm{in}^{3}=0.11725 \text { slugs } / \mathrm{ft}^{3}
\end{aligned}
$$

This is a compression ratio of ( $49.26: 1$ ) to Standard air
Back to Ideal Gas Equation, (assume no $\Delta T$ )

$$
\begin{aligned}
& P=\rho R T=\left(0.11725 \mathrm{Sl} \mathrm{mgs}^{2} / \mathrm{ft}^{3}\right)(1716 \mathrm{ftlb} / \mathrm{S} / \mathrm{ggR} \mathrm{R})(\mathrm{s} 19 \mathrm{R}) \\
& P=104,423.319 \mathrm{lb} / \mathrm{ft}^{2}=725 \mathrm{psi}-14.7 \mathrm{psi} \\
& P\left(1 / 2^{\prime \prime} \mathrm{NPS}\right)=710.5 \mathrm{psi}
\end{aligned}
$$

Need new equation!

- Need to account for Compression somehow,

PC. Launda Nichols 6/8
For ISOTHERMAL Process,

$$
\frac{P}{P}=\text { constant }
$$

For $P=2.6$ psi and $2.8 \times 10^{-3}$ slugs $/ \mathrm{ft}^{3}$

$$
\frac{P}{P}=\frac{2.6}{2.8 \times 10^{-3}}=928.57
$$

Then wing assumption of isentropic compression

$$
\frac{P_{i}}{P_{i}^{k}}=\frac{P_{f}}{P_{f^{k}}} \rightarrow P_{f}=\left(\frac{p_{f}}{p_{i}}\right)^{k} P_{i} \quad\left(k=C_{p} / C_{v}\right)
$$

where $K$ is the ratio of the specific heat at constant pressure, $C_{p}$, to the specific heat at constant volume $C_{V}$. (assuming $K=1.4$, standard air)

$$
P_{f}=\left(\frac{0.11725}{0.0028}\right)^{1.4}(17.3 p s i)=3,226.9 \text { psi ... What?! }
$$

* Note on Pg . 22 of Fluid Mechanics, Munson
"As the volume, $\forall$ is reduced by one-half, the density must double since the mass, $m=p \forall$ of the gas remains constant.
From figure EI.Cb, to keep input pressure under 50 pst. we need a compression ratio of 0.4 or less.
Trying a standard 2"NPS PVC, I.D $=2.067 \mathrm{in}$.
Max Op. Pressure $=166 \mathrm{psi} \quad($ Sch 40$)$

$$
A=\pi r^{2}=\pi(1.0335 \mathrm{in})^{2}=3.356 \mathrm{in}^{2}
$$

Or, Standard |"NPS PVC, I.D. $=1.049 \mathrm{in}, A=0.864 \mathrm{in}^{2}$ Max Op. Pressure $=270 \mathrm{psi}(\operatorname{sch} 40)$
Or, Standard $11 / 2^{\prime \prime}$ NPSPVC, $I D=1.610 \mathrm{in}, A=2.04 \mathrm{in}^{2}$

$$
\text { Max Op. Pressure }=198 \text { poi }(\text { Sch } 40)
$$

PC. Larinda Nichols 7/8
Starting with $2^{\prime \prime}$ NS PVC

$$
\rho_{2}=\frac{\rho_{1} A_{1}}{A_{2}}=\frac{\left(2.8 \times 10^{-3} \mathrm{slug} / / \mathrm{ft}^{3}\right)\left(12.73 \mathrm{in}^{2}\right)}{\left(3.356 \mathrm{in}^{2}\right)}=\frac{0.0106 \text { sump }}{\mathrm{ft}^{3}}
$$

Now, isentropic compression,

$$
P_{f}=\left(\frac{p_{f}}{\rho_{i}}\right)^{k} P_{i}=\left(\frac{0.0106}{0.0028}\right)^{1.4}(17.3 p s i)=111.85 p \text { pi }
$$

Gauge Pressure $=(111.85-14.7)$ psi $=97.15$ psi
For $11 / 2^{\prime \prime}$ NS PVC

$$
\begin{aligned}
& p_{2}=\frac{p_{1} A_{1}}{A_{2}}=\frac{\left(2.8 \times 10^{-3} \text { slog } / \mathrm{ft}^{3}\right)\left(12.73 \mathrm{in}^{2}\right)}{\left(0.864 \mathrm{in}^{2}\right)}=\frac{0.04125 \text { slugs }}{\mathrm{ft}^{3}} \\
& P_{f}=\left(\frac{0.04125}{0.0028}\right)^{1.4}(17.3 \mathrm{psi})=747.6 \mathrm{psi}
\end{aligned}
$$

Gauge Pressure $=(747.6 \mathrm{psi}-14.7 \mathrm{psi})=732.9 \mathrm{psi}$
Best Choice: $2^{\prime \prime}$ NPS Schedule 40 PVC

Summary of Pressure Calculations

1. Pressure required to accelerate water to $11.36 \mathrm{ft} / \mathrm{s}$ in 5 ft ,

$$
P_{\text {req }}=2.6 \text { (Gauge Pressure) }
$$

2. Time required to accelerate water to $11.36 \mathrm{ft} / \mathrm{s}$ in 5 ft ,

$$
t_{\mathrm{req}}=1.97 \mathrm{~s}
$$

3. Pipe Model $=4^{\prime \prime}$ NPS Schedule 40 PVC
4. Air Supply Pipe $=2^{\prime \prime}$ NPS Schedule 40 PVC
5. Pressure required to achieve 2.6 PSI in $4^{\prime \prime}$ Pipe

$$
P_{a i n}=97.15 \mathrm{psi}
$$

* No friction fores were accounted for in calculations (Friction Losses estimate don next page.)
prC. LarindaNichols 8/8
Friction has Estimates for PVC (Wiley)
$Q=A V$, assuming $11.36 \mathrm{ft} / \mathrm{s}$ Velocity
2" Pipe: $Q_{2}=\left(3.356 \mathrm{in}^{2}\right)\left(1 \mathrm{ft}^{2} / 144 \mathrm{in}^{2}\right)(11.36 \mathrm{ft} / \mathrm{s})$

$$
Q_{2}=0.265 \mathrm{ft}^{3} / \mathrm{s}\left(\frac{7.481 \mathrm{gal}}{1 \mathrm{cuft}}\right)\left(\frac{605}{\min }\right)=118.84 \mathrm{GPM}
$$

Friction loss@100GPM $=6.5 \mathrm{psi} / 100 \mathrm{ft}$
Estimated pipe length $=10 \mathrm{ft}$

$$
P_{\text {loss }}=10 \mathrm{ft}\left(\frac{6.5 \mathrm{psi}}{100 \mathrm{ft}}\right)=0.65 \mathrm{psi}
$$

So New $P \cong 98.0$ psi for $2^{\prime \prime}$ Pipe (Gauge $P$ )
4" Pipe: $Q_{4}=\left(12.73 \mathrm{in}^{2}\right)\left(1 \mathrm{ft} / 144 \mathrm{in}^{2}\right)(11.36 \mathrm{ft} / \mathrm{s})$

$$
Q_{4}=1.00 \mathrm{ft}^{3} / \mathrm{s}\left(\frac{7.481 \mathrm{cal}}{1 \text { loft }}\right)\left(\frac{60 \mathrm{~s}}{\mathrm{mmen}}\right)=448.8 \mathrm{GPM}
$$

FrictionLoss@400GPM $=3.3 \mathrm{pri} / 100 \mathrm{ft}$
Estimated pipe length $=10 \mathrm{ft}$

$$
P_{\text {loss }}=10 \mathrm{ft}\left(\frac{3.3 \mathrm{psi}}{100 \mathrm{ft}}\right)=0.33 \mathrm{psi}
$$

So New P气3.0 psi for $4^{\prime \prime}$ Pipe (Gauge P)

Pneumatic Division North America
Richland, Michigan 49083

TEC-15
Pipe Air Flow
ISSUED: January, 1999
Supersedes: June, 1998

## Pipe Air Flow

The following pages contain 6 sets of curves for schedule 40 pipe that can be used to help select the appropriate pipe size for pneumatic systems, or given a system, allow system performance to be estimated.

Generally accepted practice for sizing piping for pneumatic systems is to use a pressure drop of $10 \%$ of gage for nominal pipe sizes up to and including $1 / 2^{\prime \prime}$, and $5 \%$ of gage for nominal pipe sizes of $3 / 4^{\prime \prime}$ and larger. The following curves allow the use of these guidelines for selecting piping sizes and include other pressure drop percentages for evaluating existing systems. Generally, curves of this type are shown only for 100 feet pipe lengths, but theoretic calculations show the curves for 10 feet are also valid.

Below is a listing of the charts involved with their identification:

| Pipe Size Range | Pressure Drop <br> (Percentage of Inlet Gage Pressure) |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
| :--- | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $1 / 8^{\prime \prime}-1 / 2^{\prime \prime}$ | 5 | 10 | 15 | 5 | 10 | 15 |  |  |  |  |  |  |  |  |  |
| $3 / 4^{\prime \prime}-3^{\prime \prime}$ | 2.5 | 5 | 7.5 | 2.5 | 5 | 7.5 |  |  |  |  |  |  |  |  |  |
| Pipe Length (Feet) | 100 | 100 | 100 | 10 | 10 | 10 |  |  |  |  |  |  |  |  |  |
| Chart | A | B | C | D | E | F |  |  |  |  |  |  |  |  |  |
| Generally accepted practice. |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |

Perhaps the best way to explain the use of these curves is by example.

## Example 1

Given a system with desired airflow of 700 SCFM and a supply pressure of 60 PSIG and a header length of 100 feet, what size pipe should be used? The generally accepted practice of $10 \%$ pressure drop for pipes up to $1 / 2^{\prime \prime}$ and $5 \%$ for $3 / 4^{\prime \prime}$ and larger should be used.

The above table indicates, Chart B should be used (Step 1). Along the bottom horizontal axis locate the 60 PSIG vertical line (Step 2). On the left vertical axis locate the 700 SCFM horizontal line (Step 3). Follow both of these lines to the point of intersection.

This occurs between the sloping lines for the $2^{\prime \prime}$ and $2-1 / 2^{\prime \prime}$ pipes. The larger pipe size ( $2-1 / 2^{\prime \prime}$ ) should be selected (Step 4). Further evaluation of this chart shows that for the conditions given, the pipe will flow over 800 SCFM at 60 PSIG inlet and 3 PSI (5\%) pressure drop. (The intersection of the 60 PSIG primary pressure line and the $2-1 / 2^{\prime \prime}$ pipe size line).

Further uses of the curves would be to compare the size pipe required at the other sets of pressure drops for 100 feet of pipe length. Using Chart A shows that if a more conservative pressure drop were used, the pipe size would increase to $3^{\prime \prime}$. Using Chart C shows that if a more aggressive pressure drop were allowed, perhaps a $2^{\prime \prime}$ pipe could be used.

Using the curves for 10 feet of pipe length, it can be seen because of the shorter length, much smaller pipe diameters could be used than if the length were at 100 feet.

These curves should only be used as general guidelines for selecting piping systems. Also, these curves are based on using schedule 40 steel pipe. Different types of plumbing with different internal roughness will have different results. If more detailed or precise information is required, the system should be designed by a competent professional.

## Flow of Air Through a Pipe



## APPENDIX H: Modular Sectional Model Drawings

| Sectional Model |  |  |  |
| :--- | :---: | :---: | :---: |
| S.no. Description Quantity Drawing No. <br> 1 Inclined Inlet   <br>  $25^{\circ}$ 1 $2-10$ <br>  $35^{\circ}$ 1  <br>  $45^{\circ}$ 1  <br> 2 Gate Channel 1 $11-13$ <br> 3 Outlet box 1 $14-17$ <br> 4 Additional Length component 2  <br>  $0.2^{\prime}$ 4 $18-26$ <br>  $0.4^{\prime}$ 2  <br> 5 $1^{\prime}$ Flanges 22 |  |  |  |
















DRAWING NO.: 23
















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